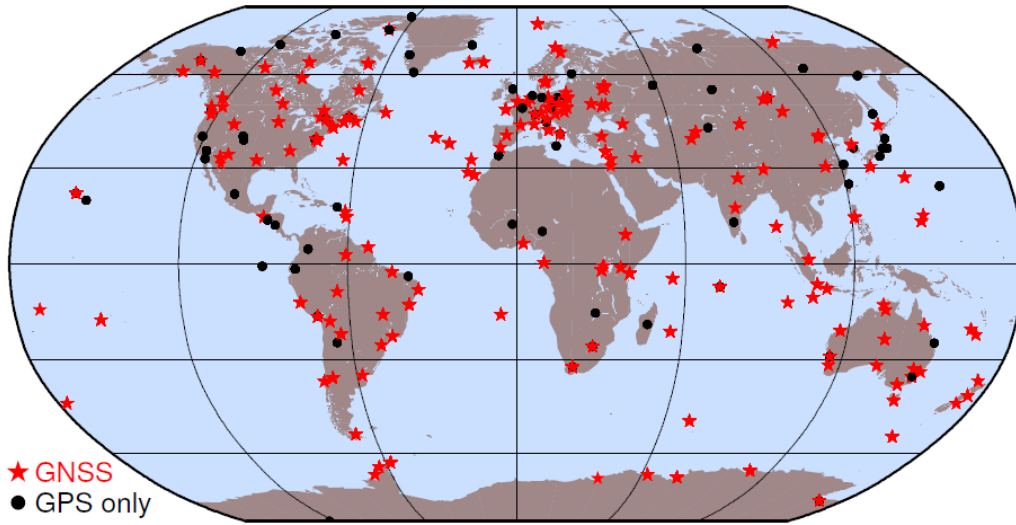
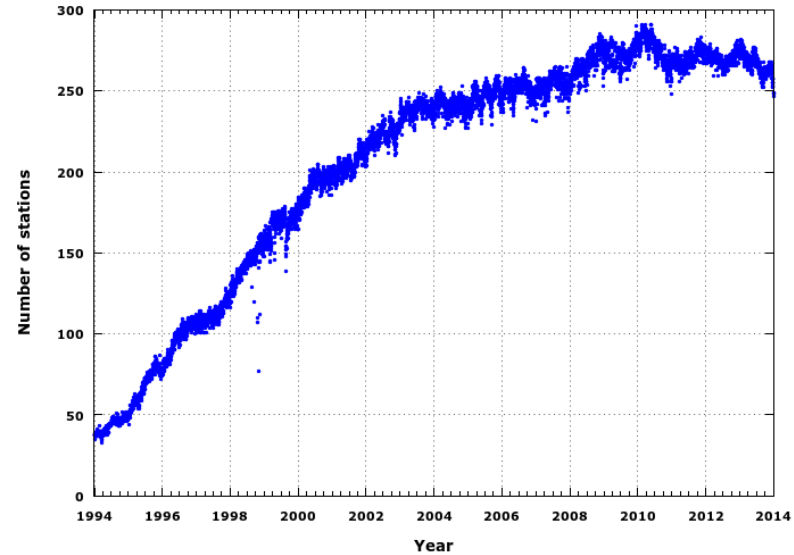


Reference Frame Products

Andreja Susnik, Rolf Dach, Andrea Maier,
Daniel Arnold

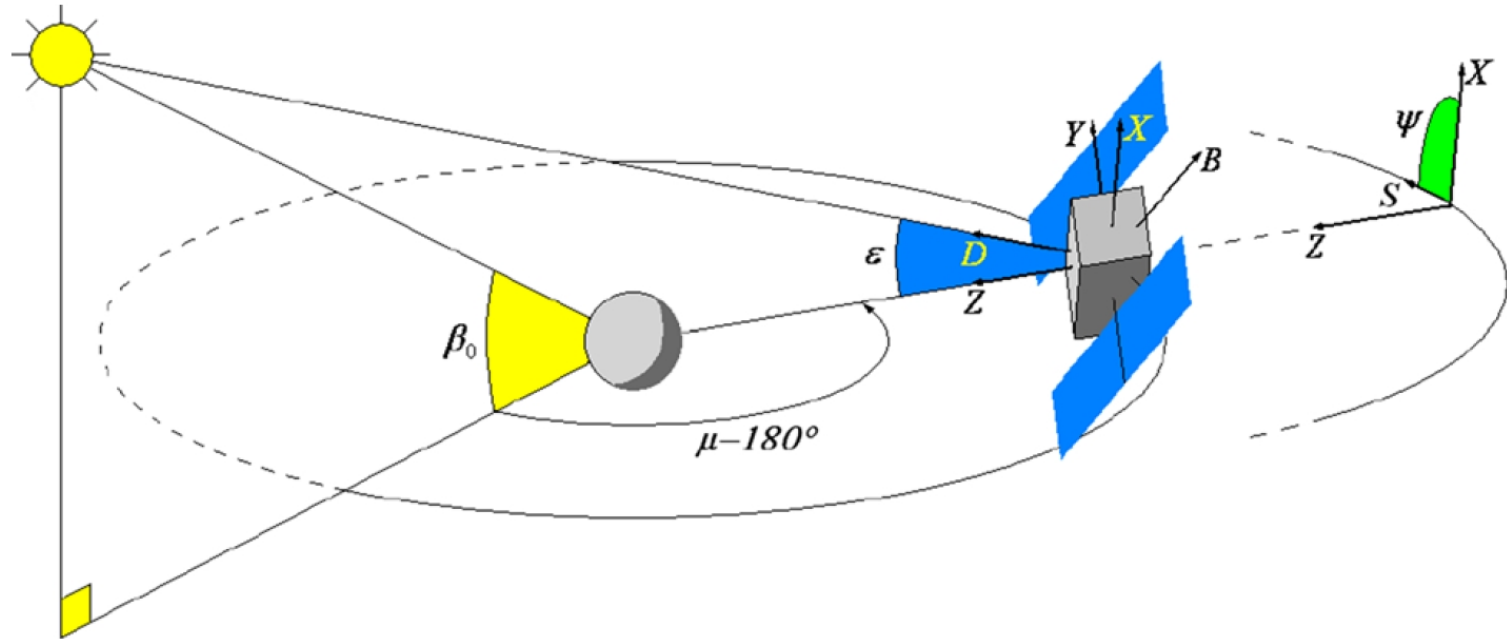
Introduction

- for consistent series of GNSS satellite clock corrections, orbits, Earth rotation parameters and station coordinates were homogeneously processed for the interval between 2000 to the end of 2014, using the latest development version of the Bernese GNSS software



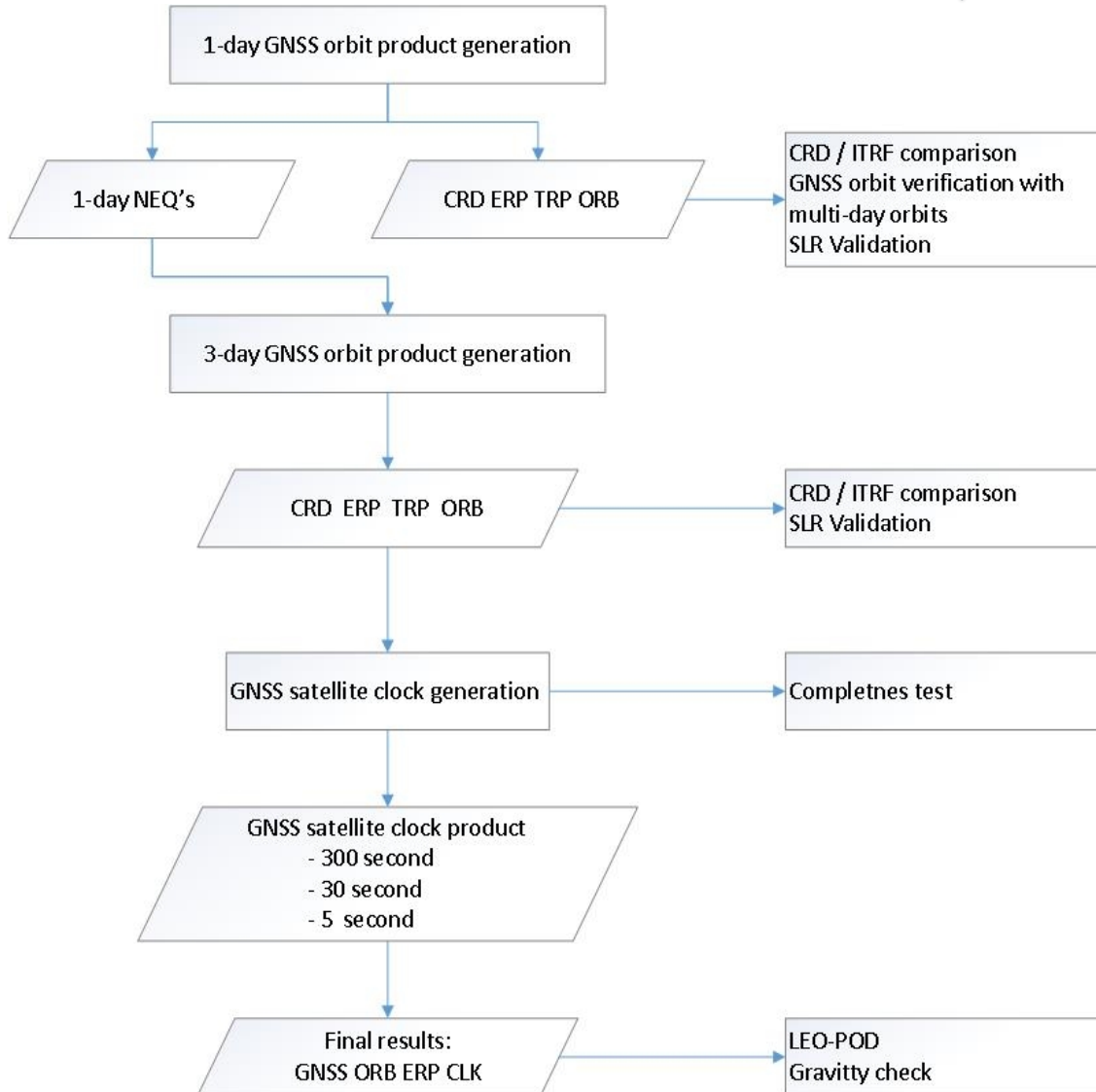
Introduction

- in order to provide the best reference frame products, the latest GNSS orbit model was used (Arnold et al., 2015) for the full period of reprocessing efforts



| | Parameters estimated in | | |
|---------------|-------------------------|----------|-----------------|
| | D | Y | B |
| Original ECOM | constant | constant | constant, 1-cpr |
| Extended ECOM | constant, 2-cpr, 4-cpr | constant | constant, 1-cpr |

Processing Scheme



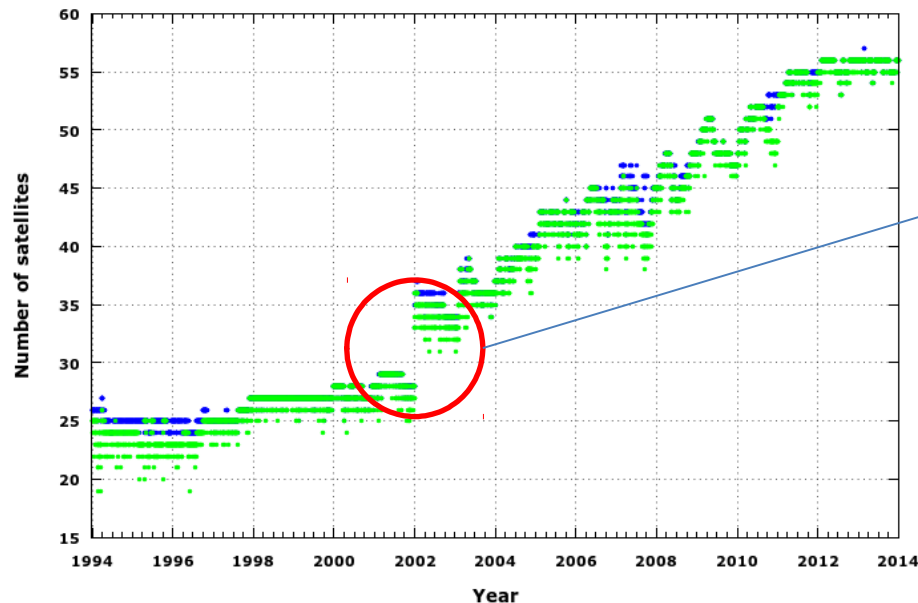
- as the basis for the GRACE orbit determination, based on its onboard GPS receivers, GNSS satellite orbits, Earth rotation parameters (ERP's) and GNSS satellite clock corrections (at 30 and 5 s sampling rate), attached to the IGB08 reference frame are computed
- during the processing several quality control steps were established

Processing Scheme

1-day GNSS orbit product generation

- original GNSS observations in RINEX files (RINEX2 format)
- as a priori orbit information the results from repro02 were used and completed with alternative sources (i.e., broadcast orbits)
- based on repro02, all known RINEX inconsistencies are corrected
- full pre-processing and ambiguity resolution scheme is applied (DD, receiver and satellite clocks are pre-eliminated)

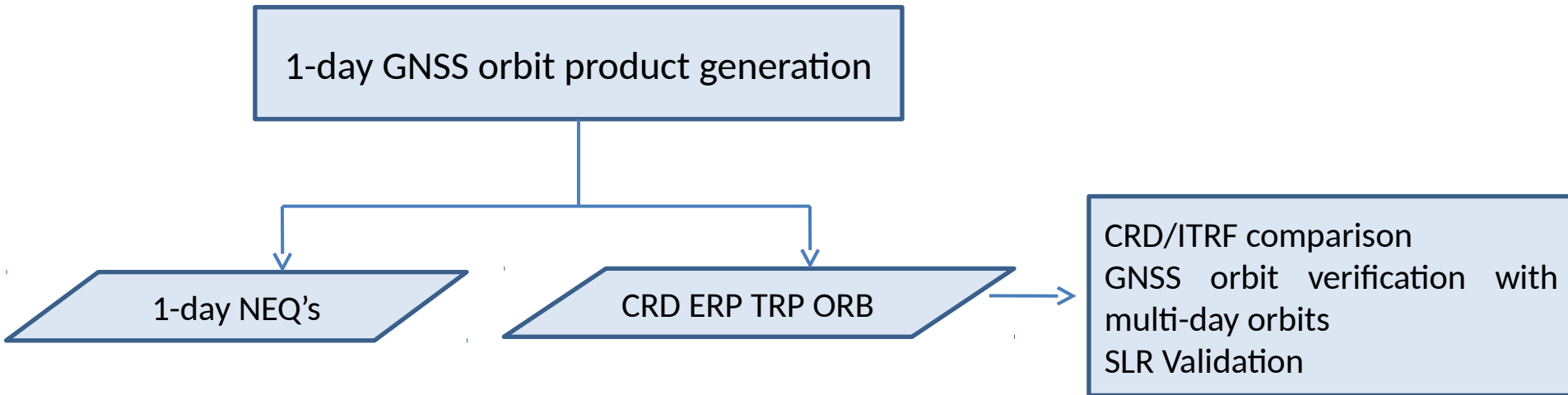
Repro 03 • Repro 02 •



Inclusion of GLONASS

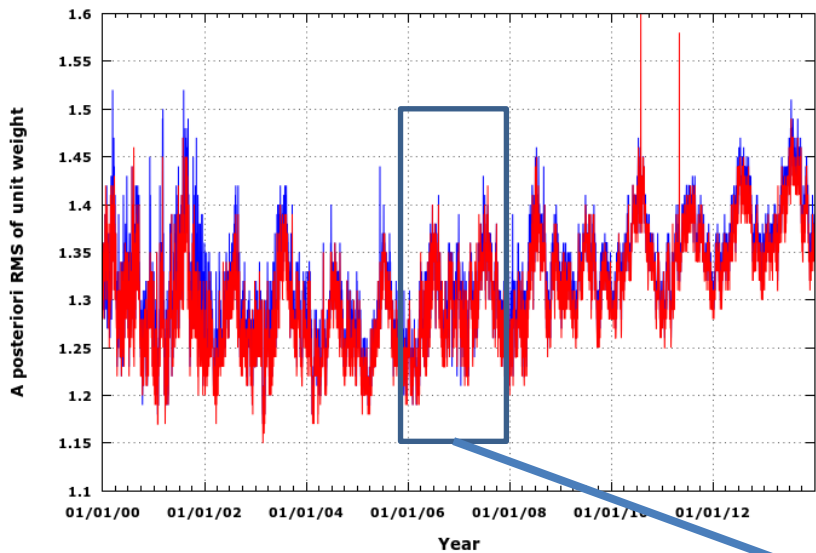


Processing Scheme

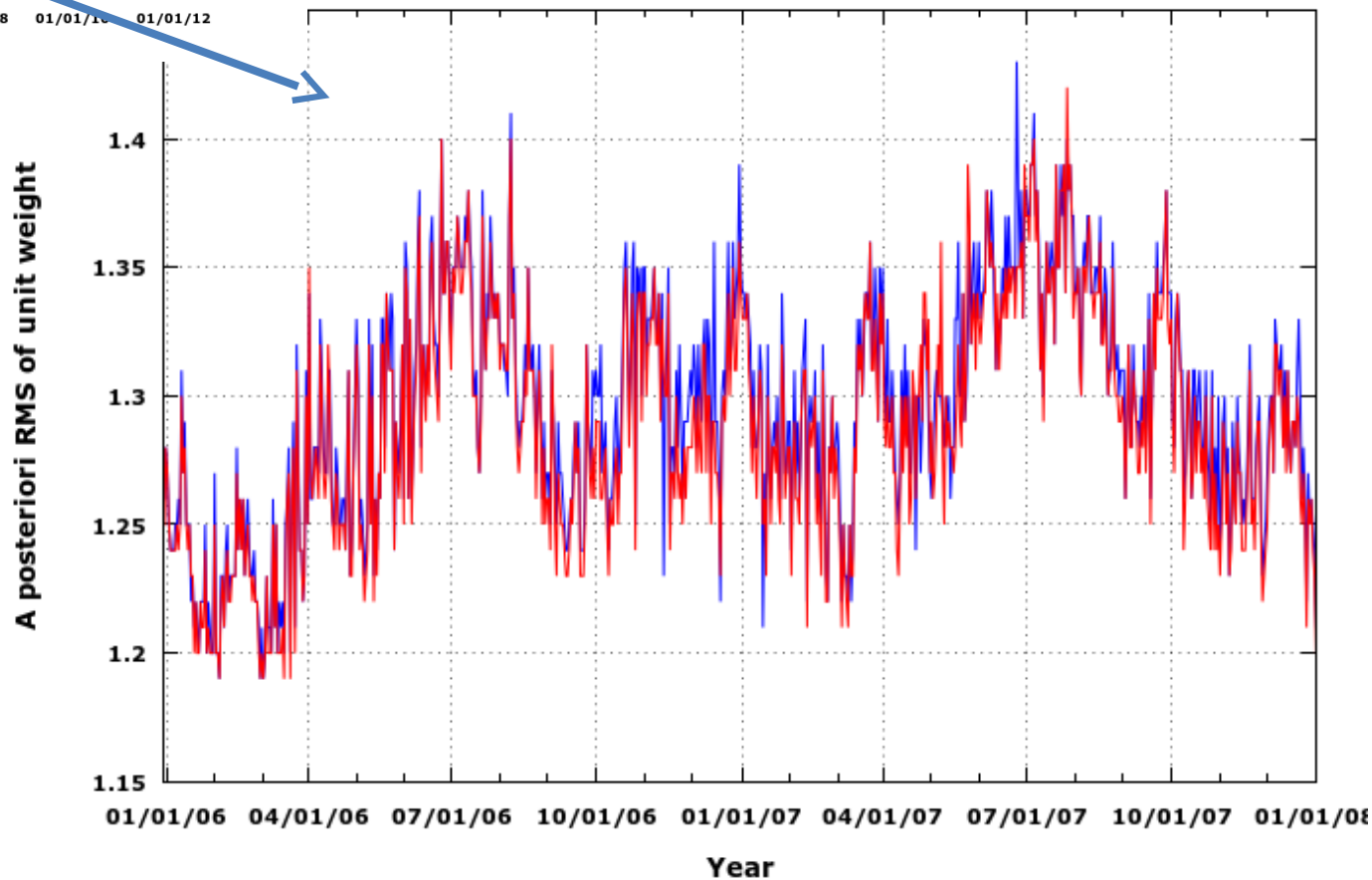


- main product are 1-day NEQ's, containing GNSS satellite orbit parameters, ERP's, coordinates and troposphere zenith path delay parameters

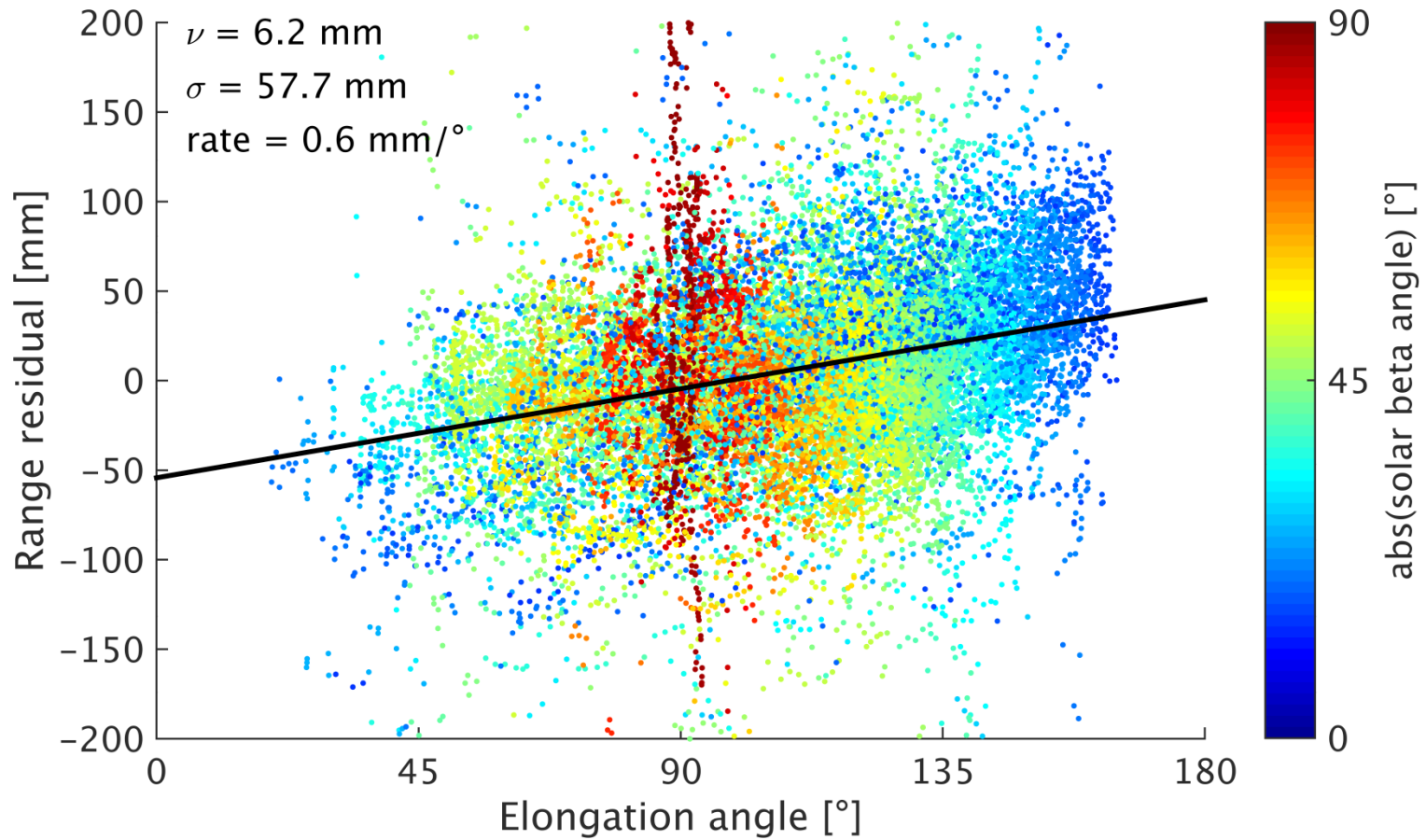
Repro 02 ———— Repro 03 ————



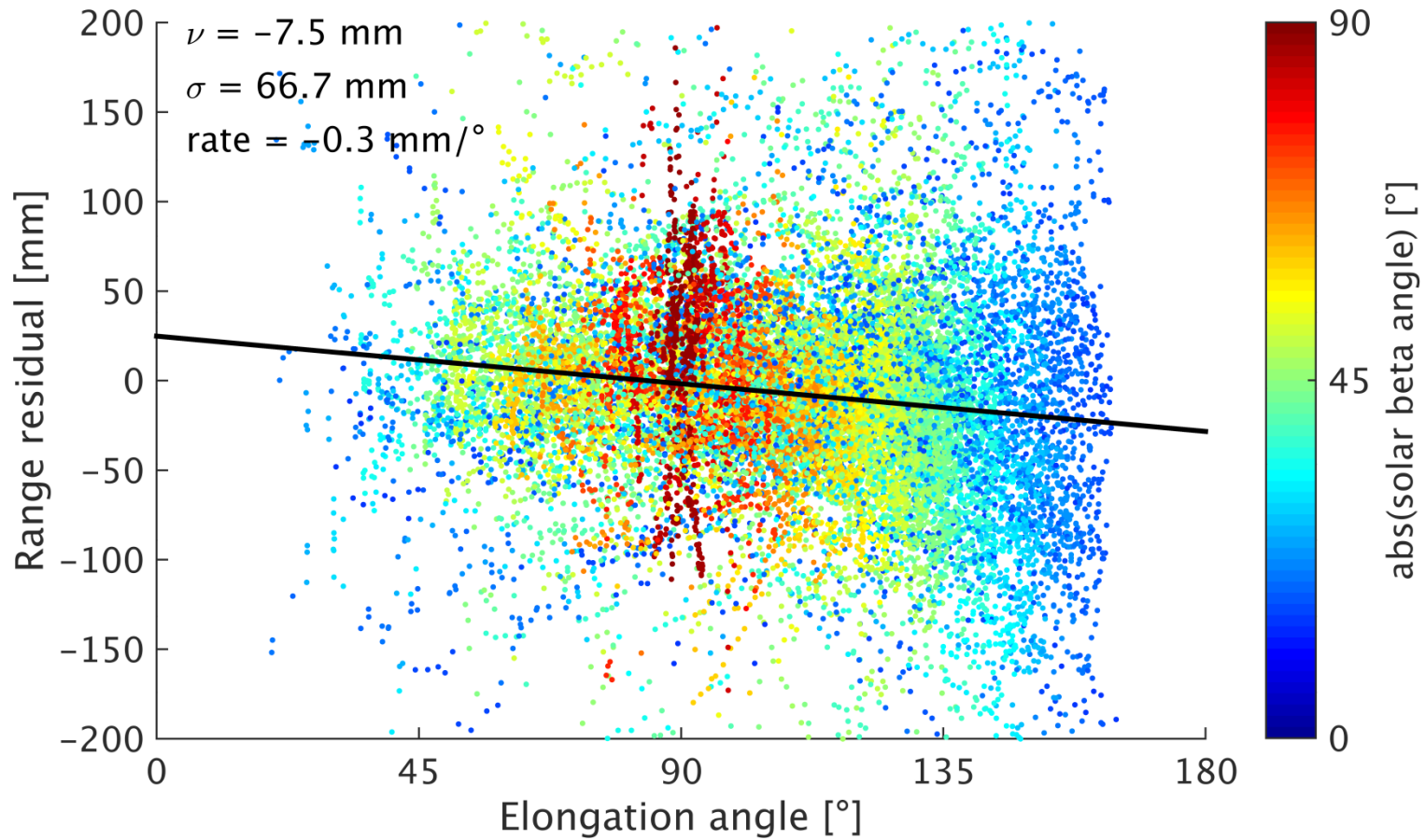
Repro 02 ———— Repro 03 ————



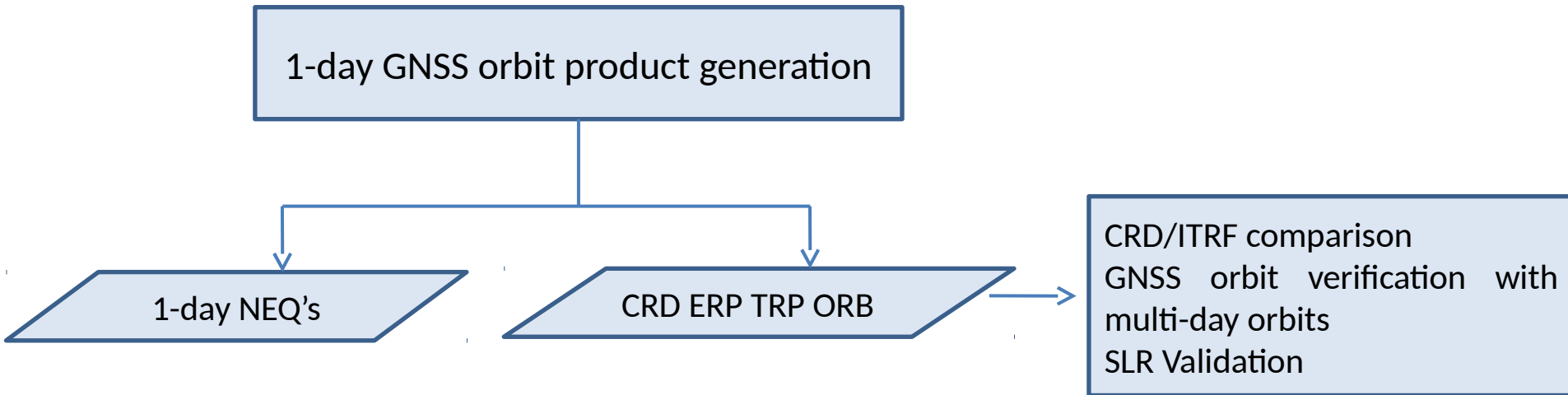
SLR validation



SLR validation

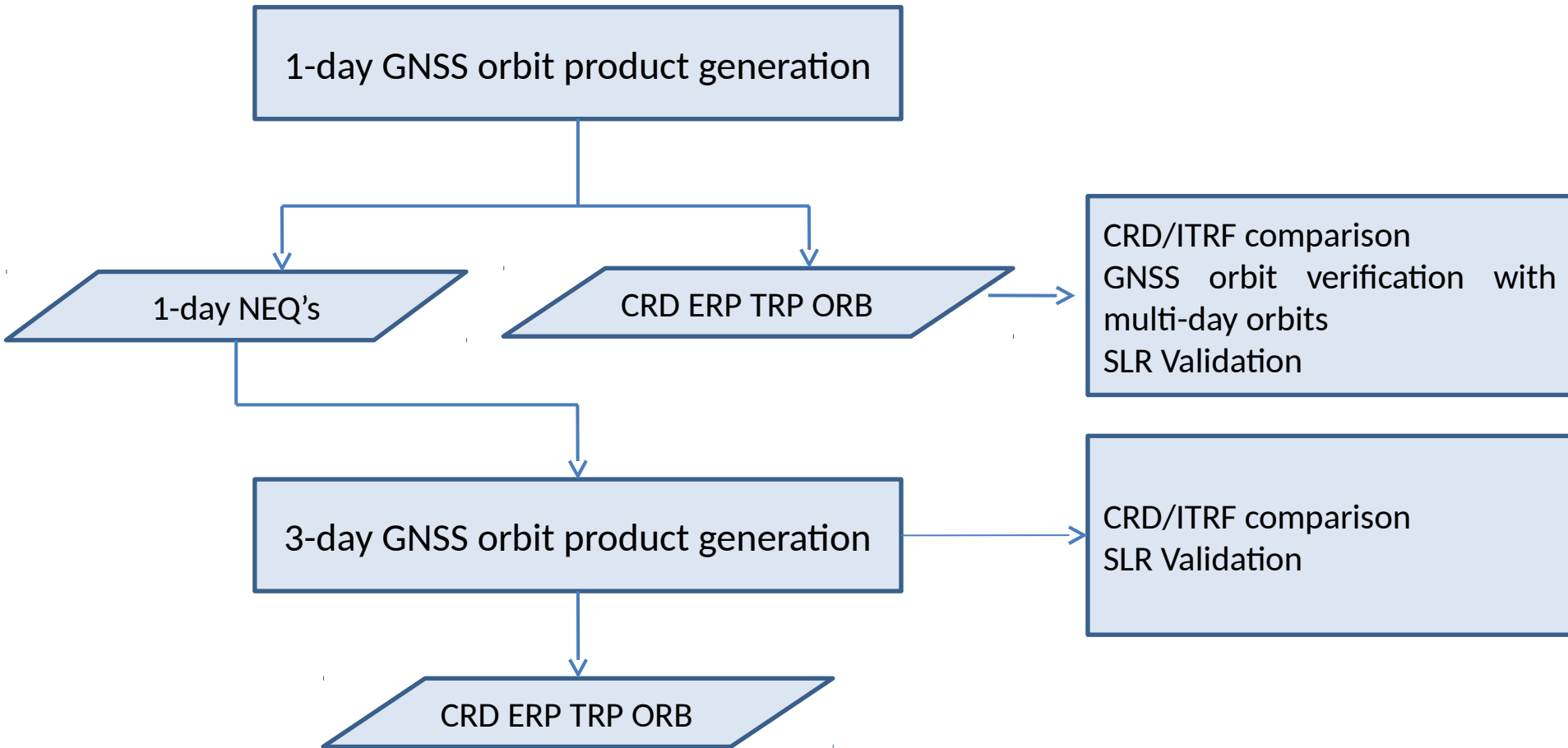


Processing Scheme



- in next step the consistency of three subsequent one-day orbits is verified
- additionally, a verification of the station related parameters is performed
- all remaining RINEX inconsistencies are corrected
- after these preparatory steps three subsequent NEQs are combined and solved to a three-day long-arc solution

Processing Scheme



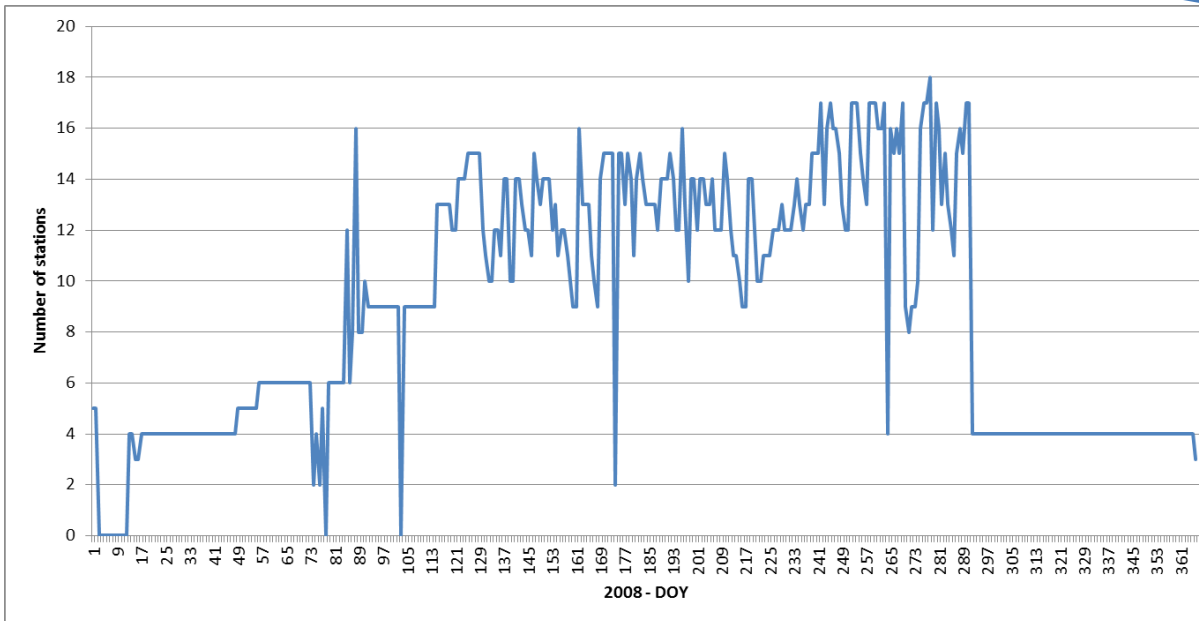
Processing Scheme



- in the next step, GNSS satellite clock generation is performed
- the procedure is based on Bock et al. (2009) and it has been extended in the frame of the project to a GPS and GLONASS combined processing scheme

Processing Scheme

- in the next step, GNSS satellite clock generation is performed
- the procedure is based on Bock et al. (2009) and it has been extended in the frame of the project to a GPS and GLONASS combined processing scheme



- due to very sparse availability of GLONASS data, we have included GLONASS from 2008 onwards in 30 s satellite corrections, while before 2009 GLONASS is completely excluded

GNSS satellite clock generation



Clean code + phase
Smooth code

Pre-processing

- Screening
station-by -station
1. Phase only
 2. Code only
 3. Phase + Code

screening of post-fit
residuals

low-rate clock solution

3 global clusters
45 stations/cluster



combination +merging



5- min clocks

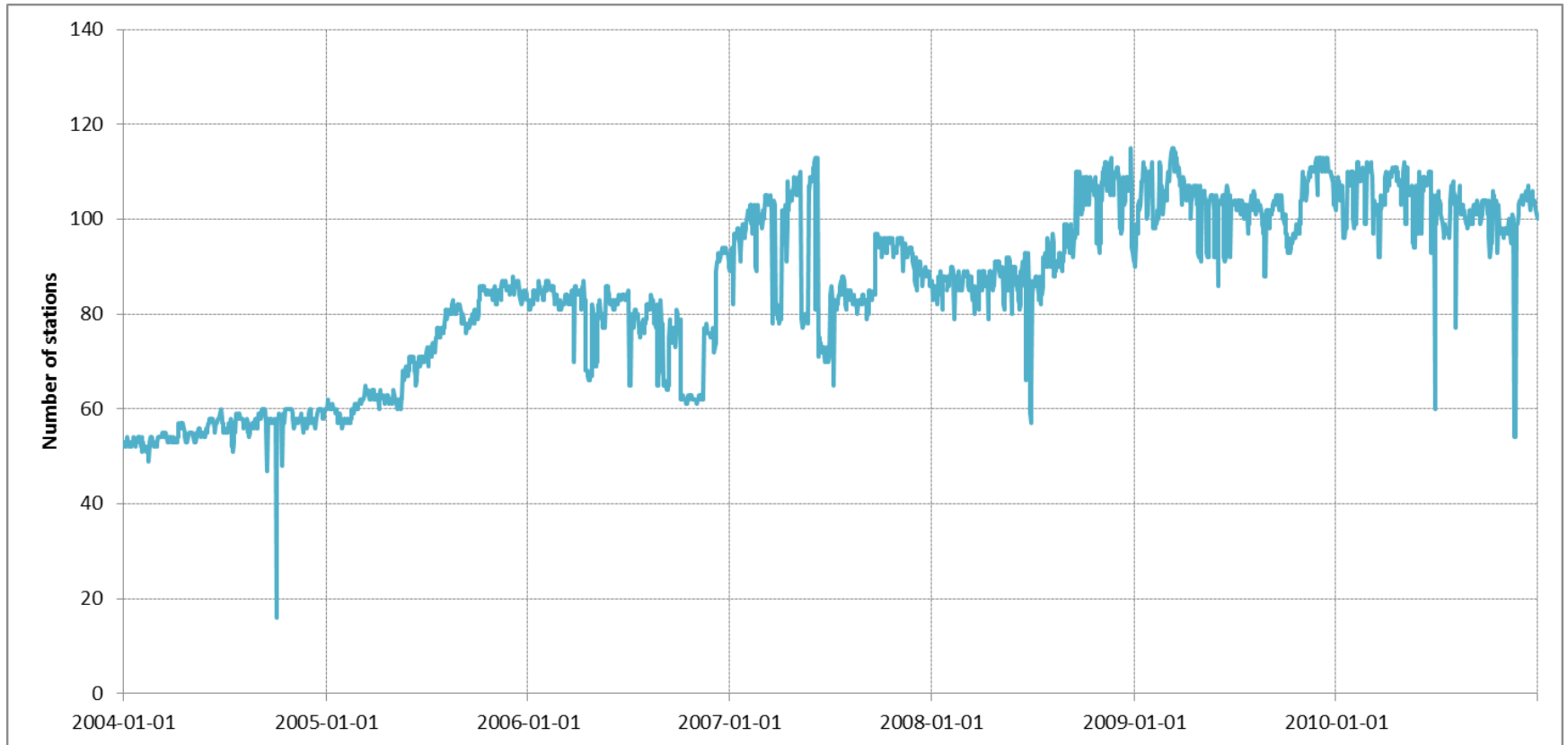
EHRI from 5 min to 30 s

high-rate clock solution

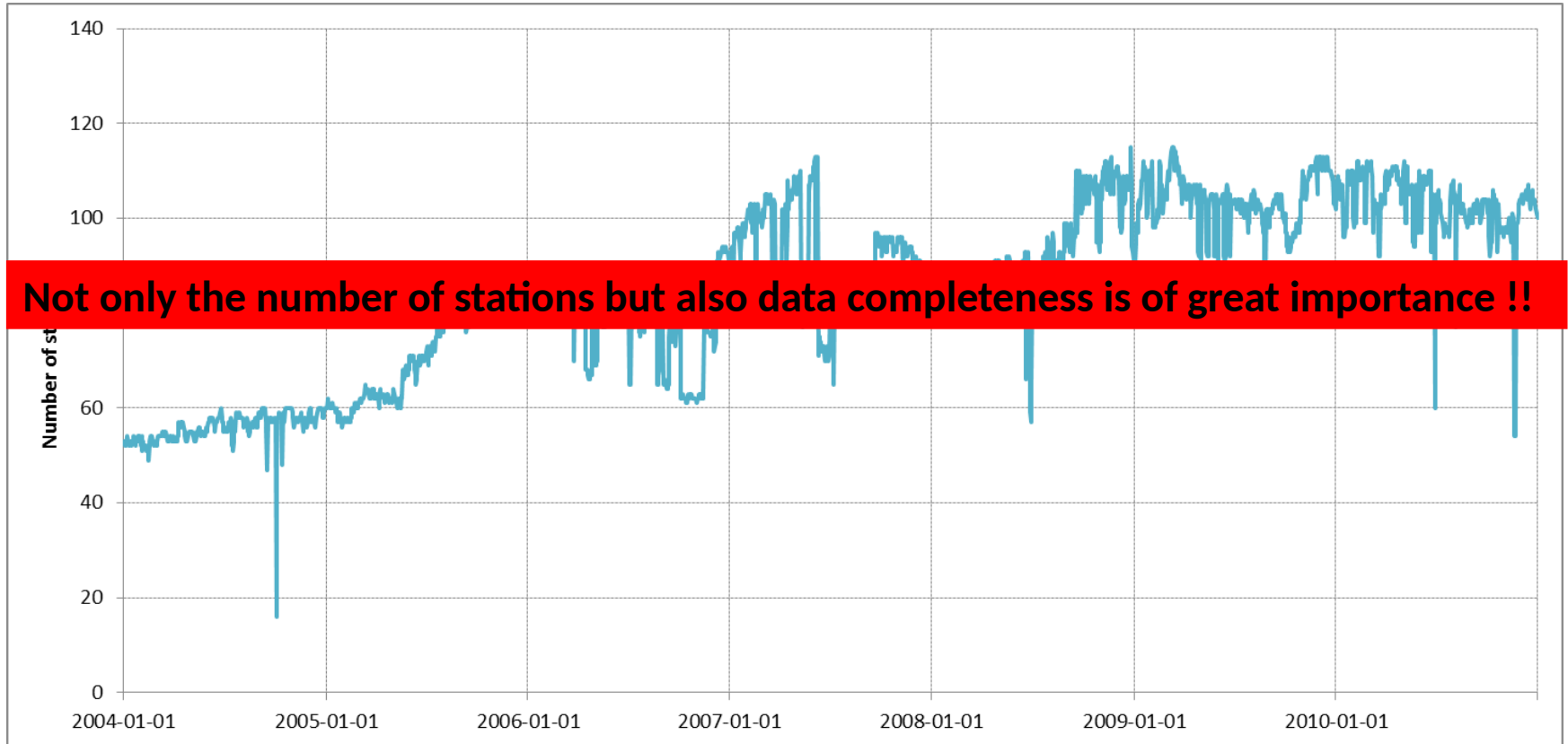
EHRI from 30 s min to 5 s



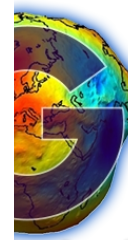
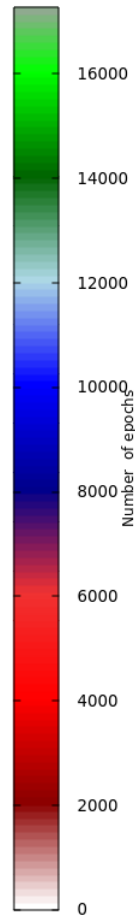
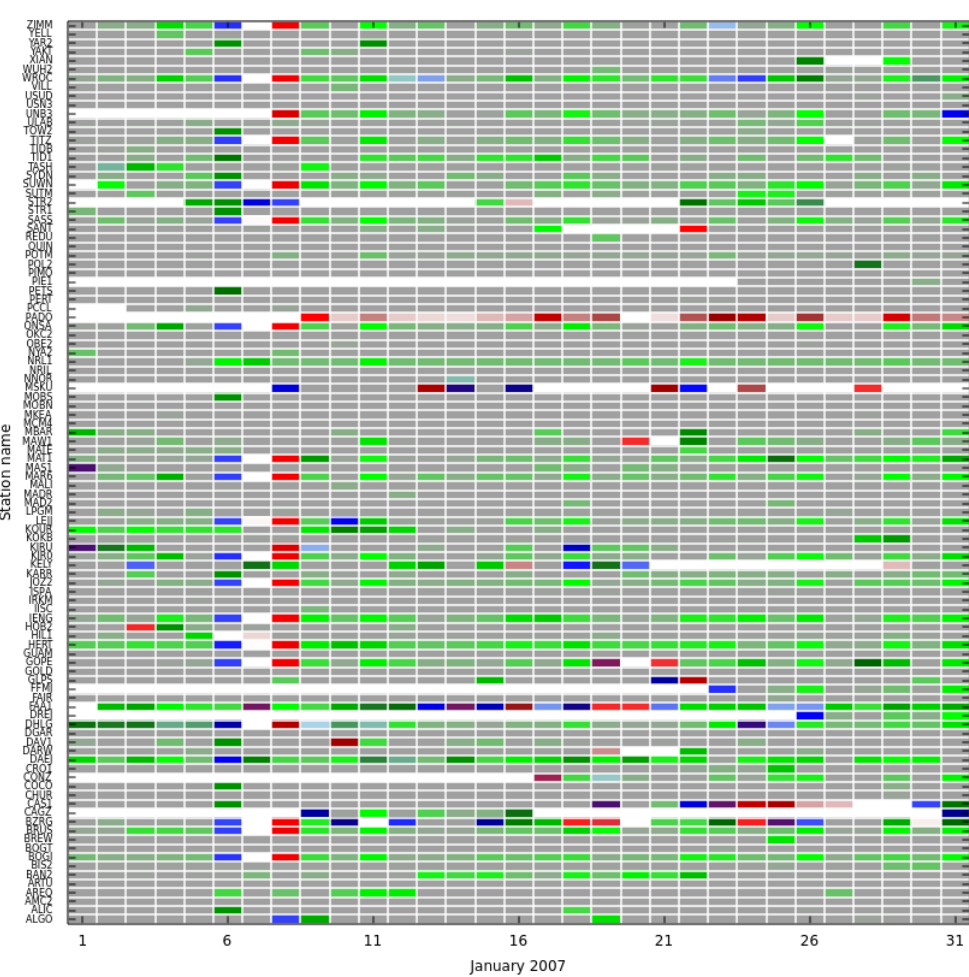
HORIZON 2020



High-rate RINEX2 station data availability at AIUB datapool, for the period between 2004-2010



High-rate RINEX2 station data availability at AIUB datapool, for the period between 2004-2010



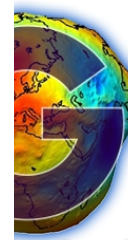
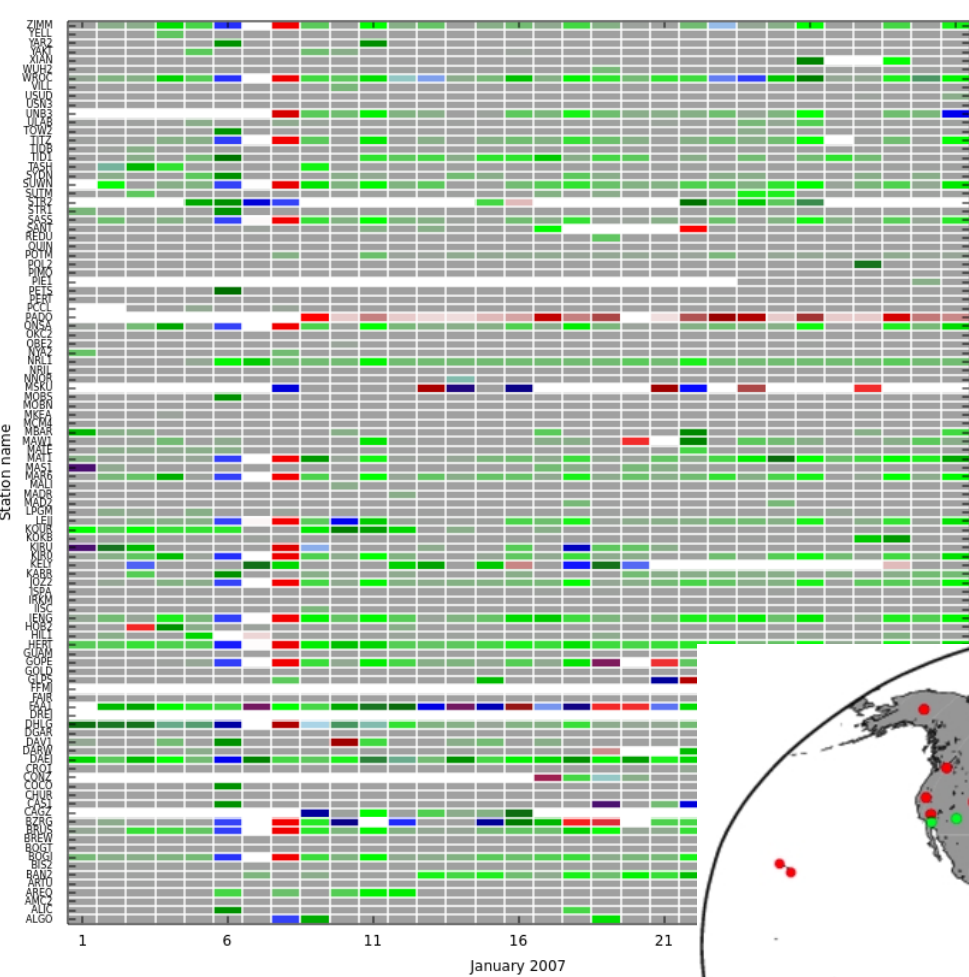
OSIEM

ty Service for Improved Emergency Management

grey color present full completeness



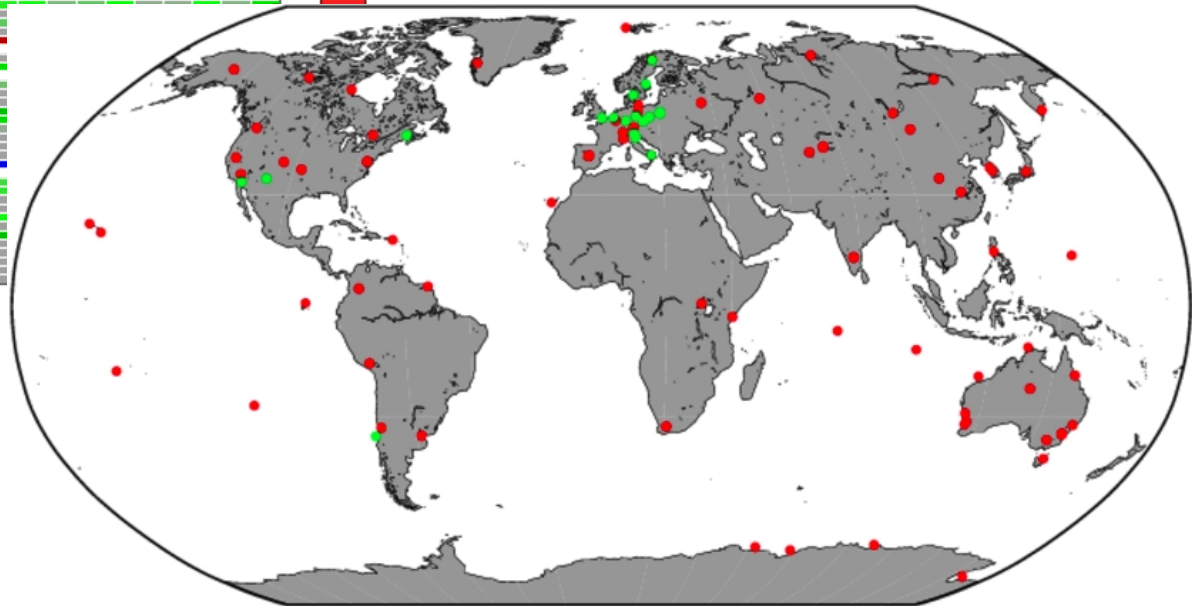
HORIZON 2020



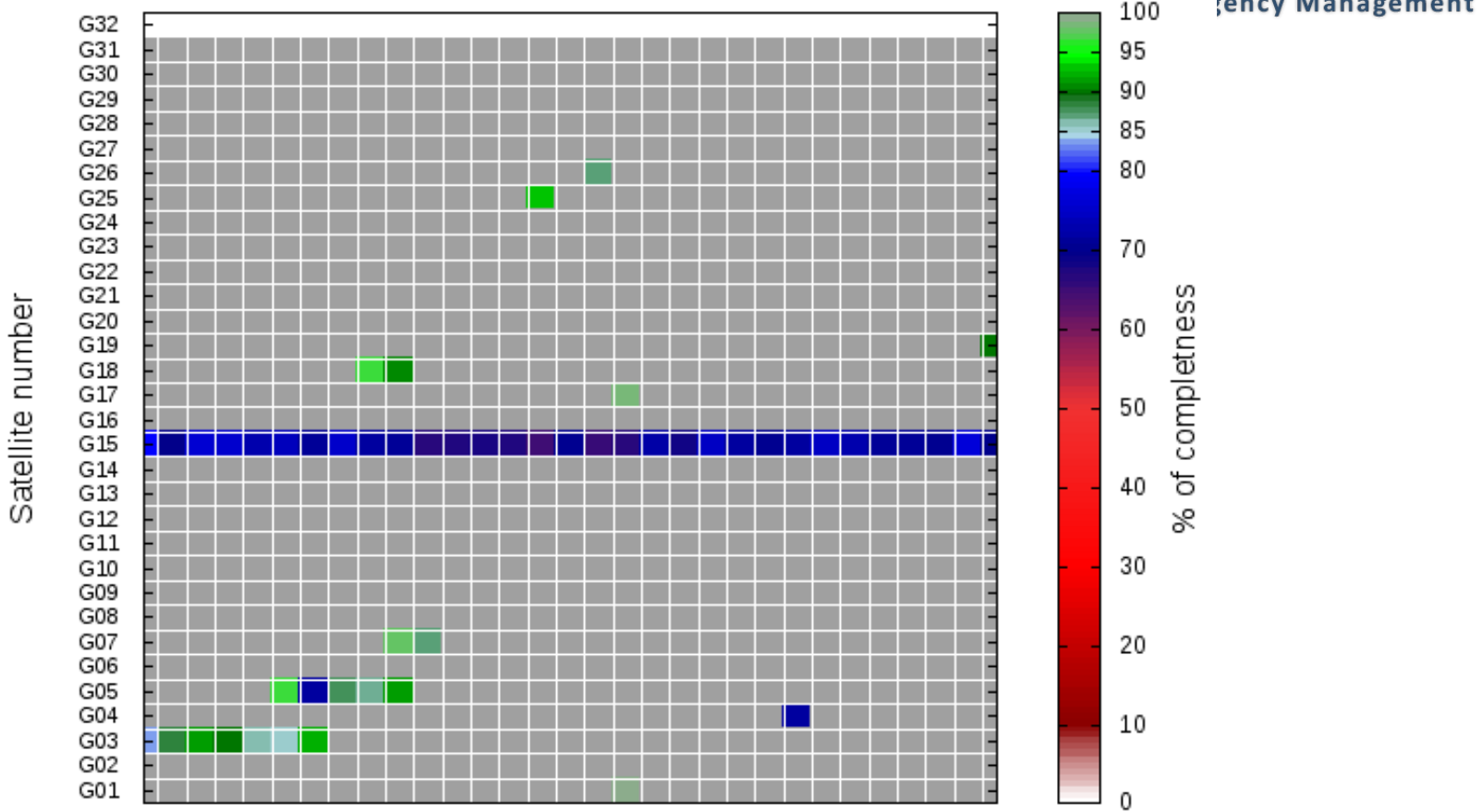
ESIEM

ty Service for Improved Emergency Management

7 Januar 2007
27 Januar 2007



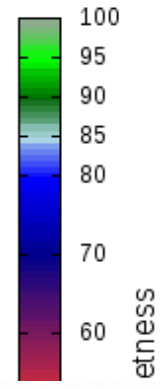
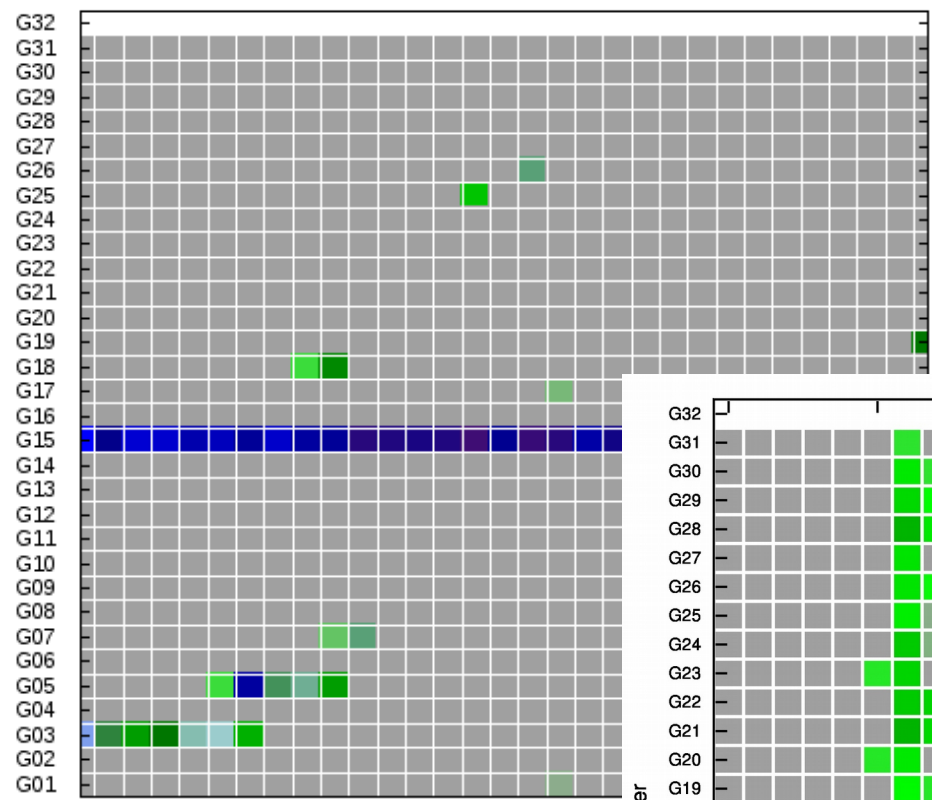
grey color present full completeness



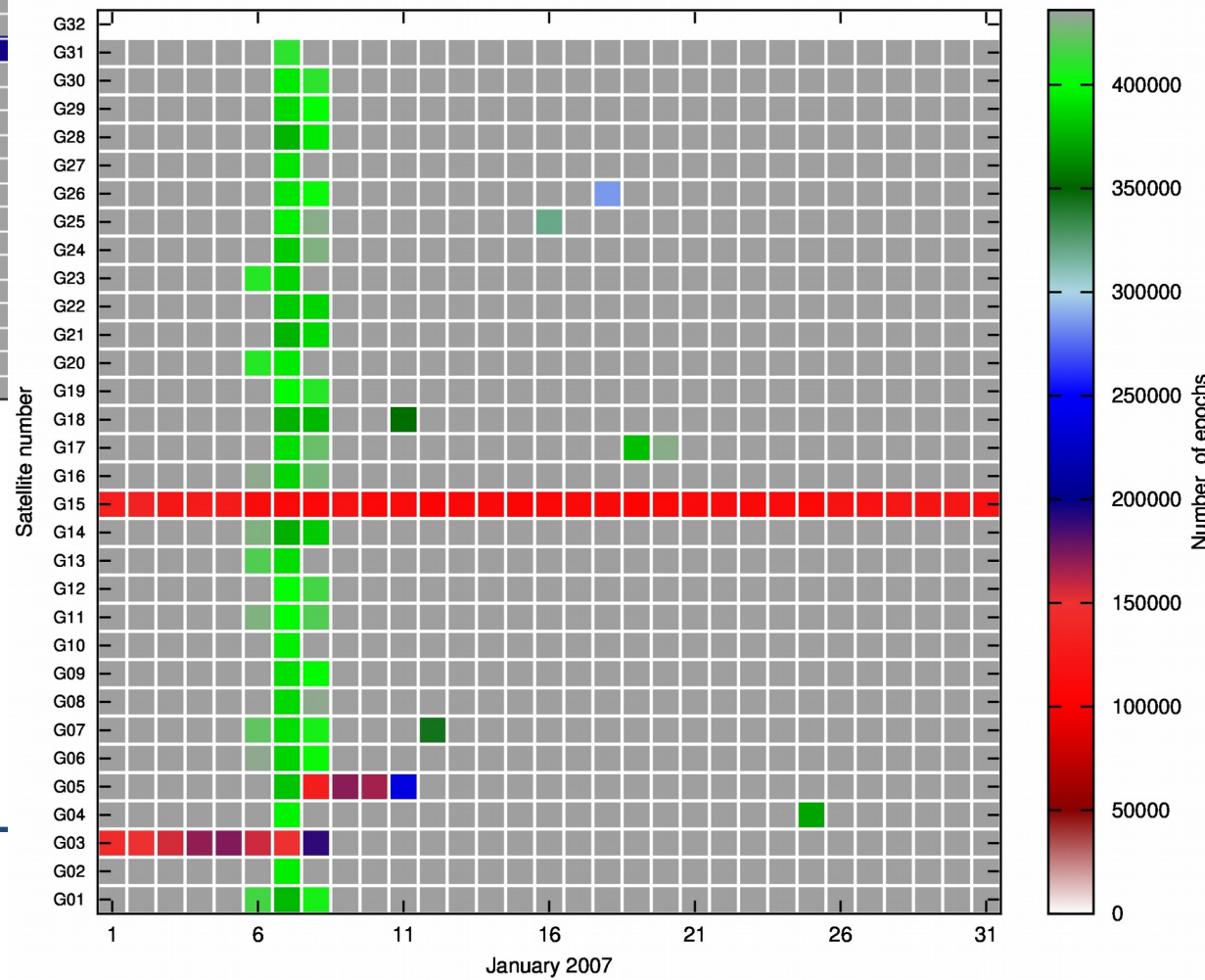
grey color present full completeness

January 2007

Satellite number



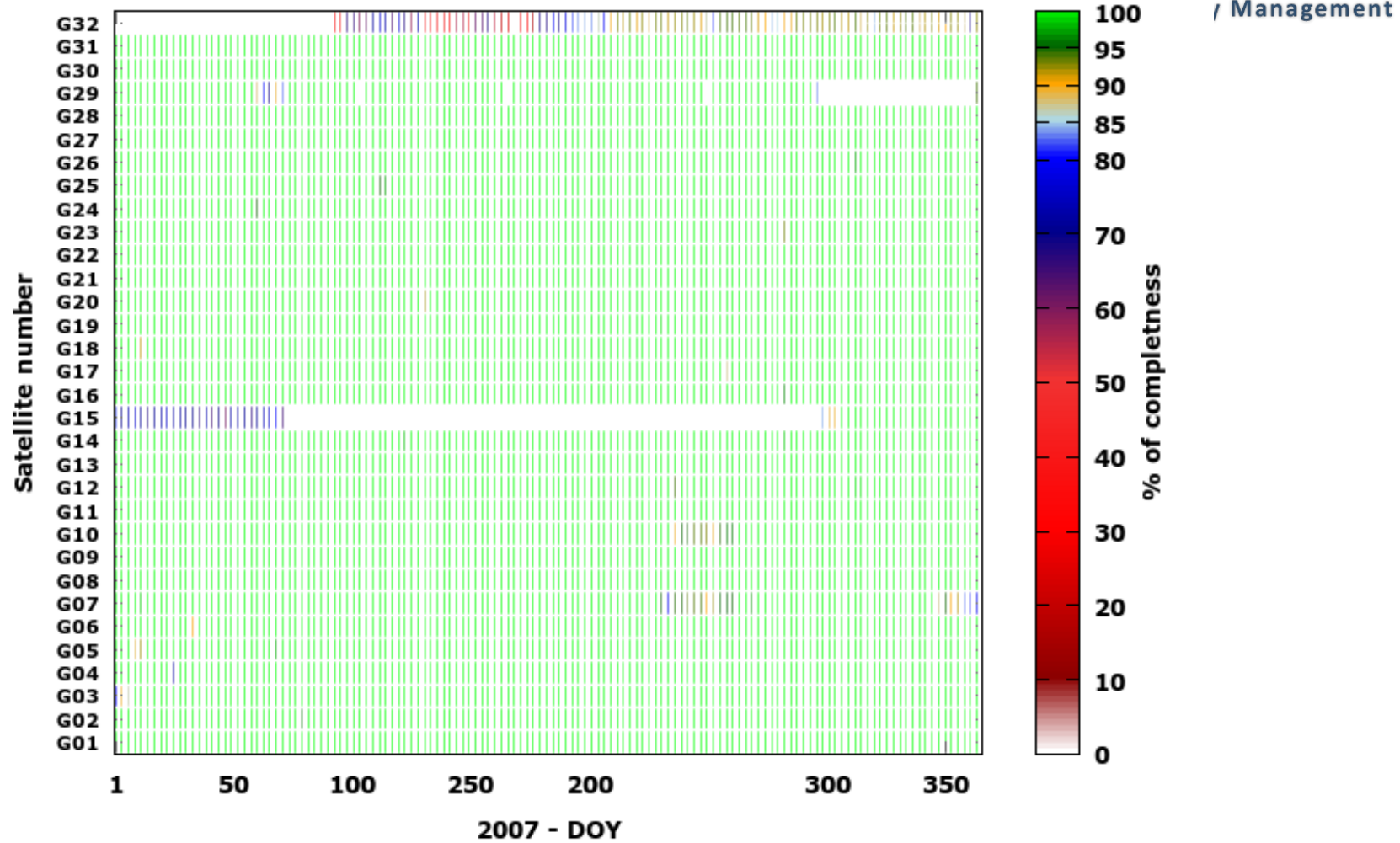
January 2007



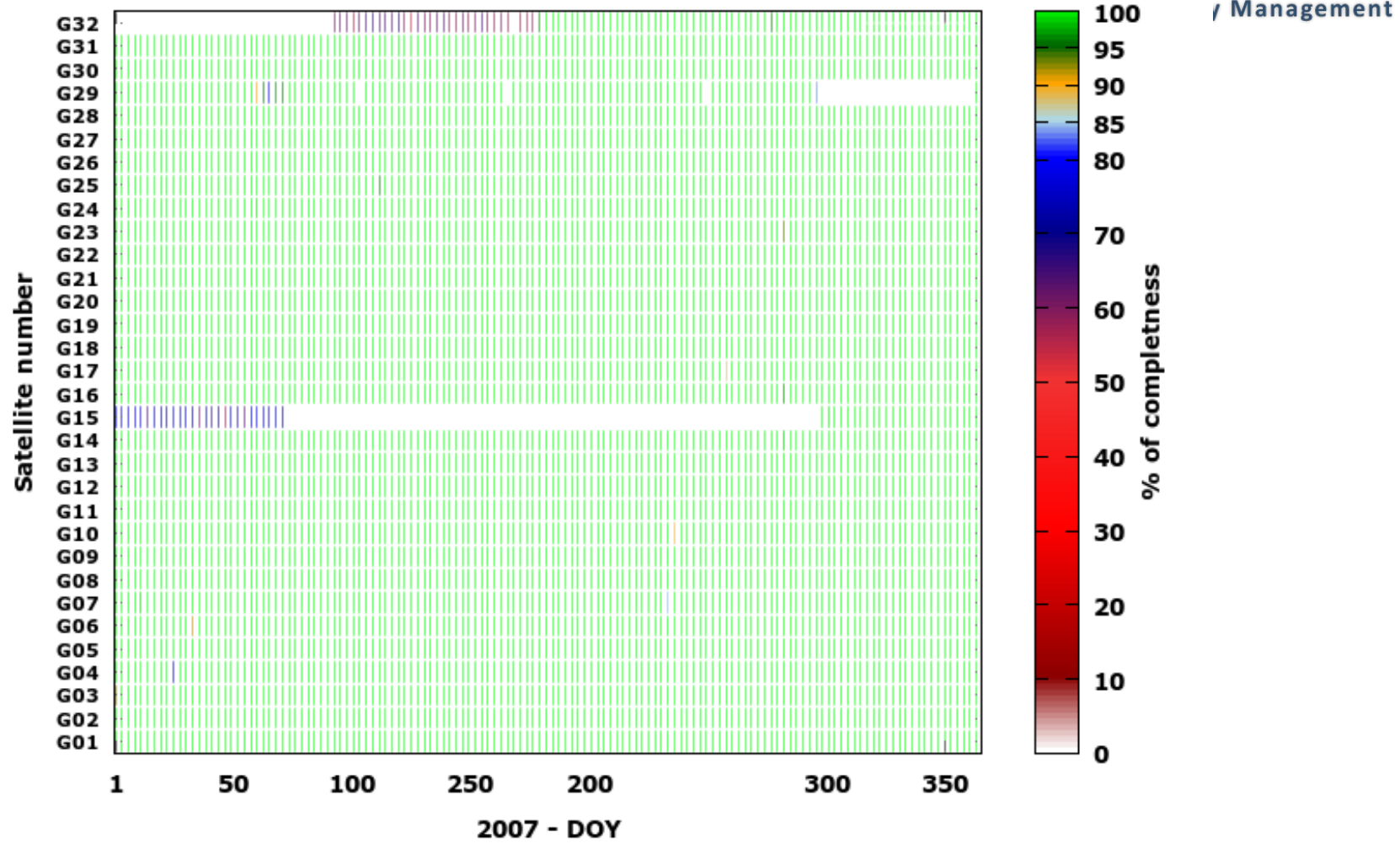
grey color present full completeness

January 2007

EOSIEM

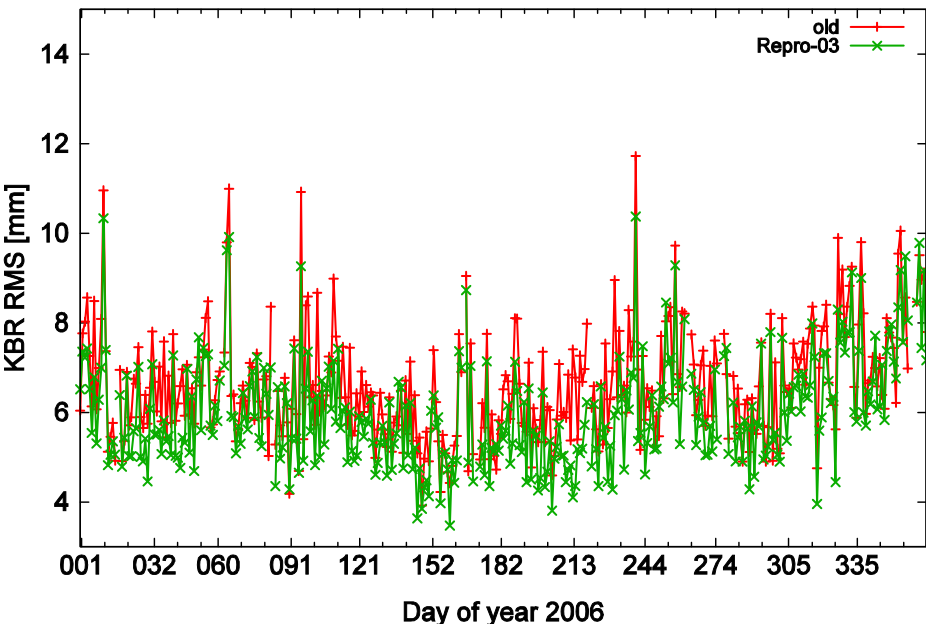


EOSIEM

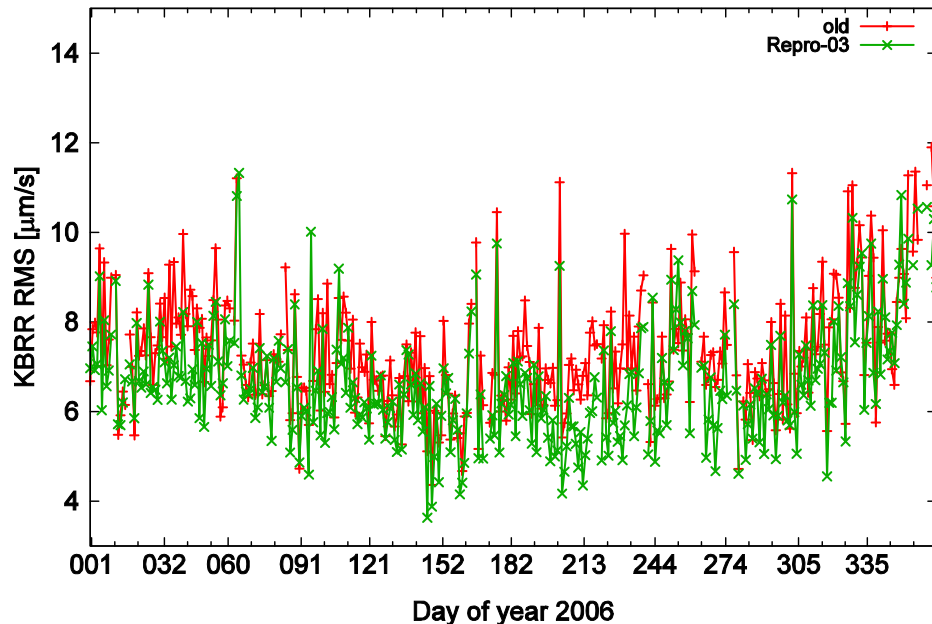


Validation by GRACE Orbit Determination

K-band validation of GRACE red.-dyn. orbits

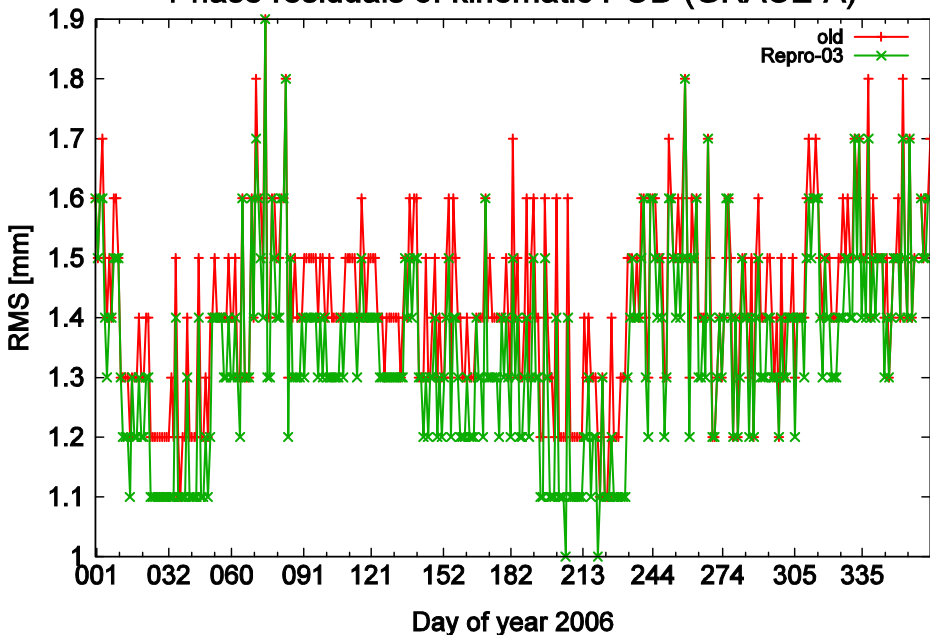


K-band validation of GRACE red.-dyn. orbits

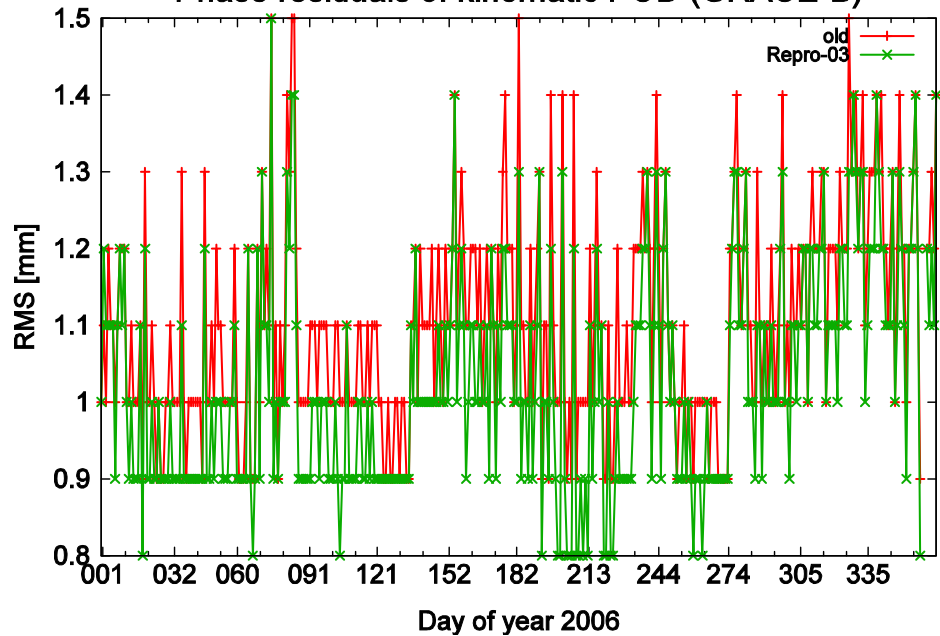


Validation by GRACE Orbit Determination

Phase residuals of kinematic POD (GRACE-A)

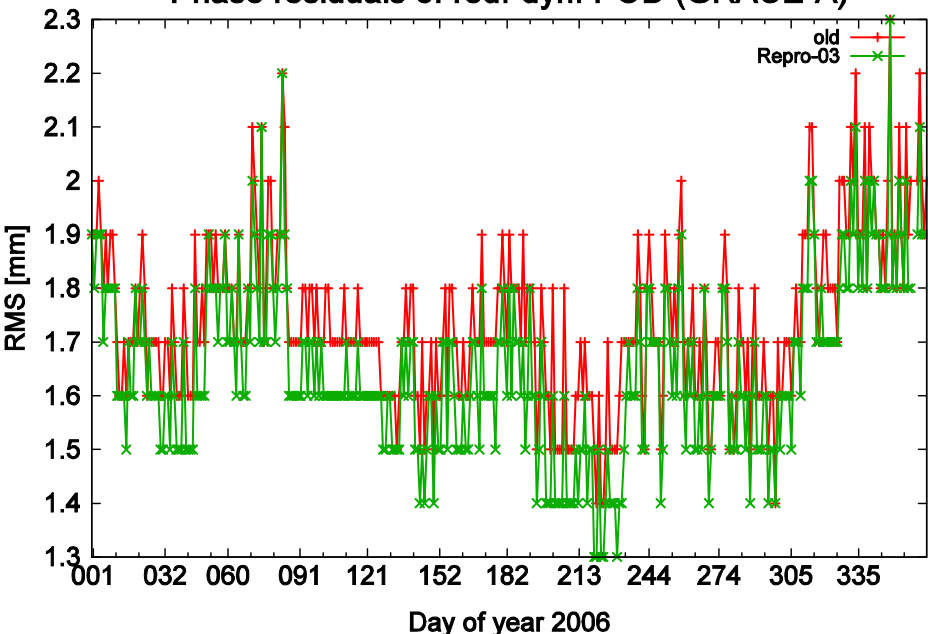


Phase residuals of kinematic POD (GRACE-B)

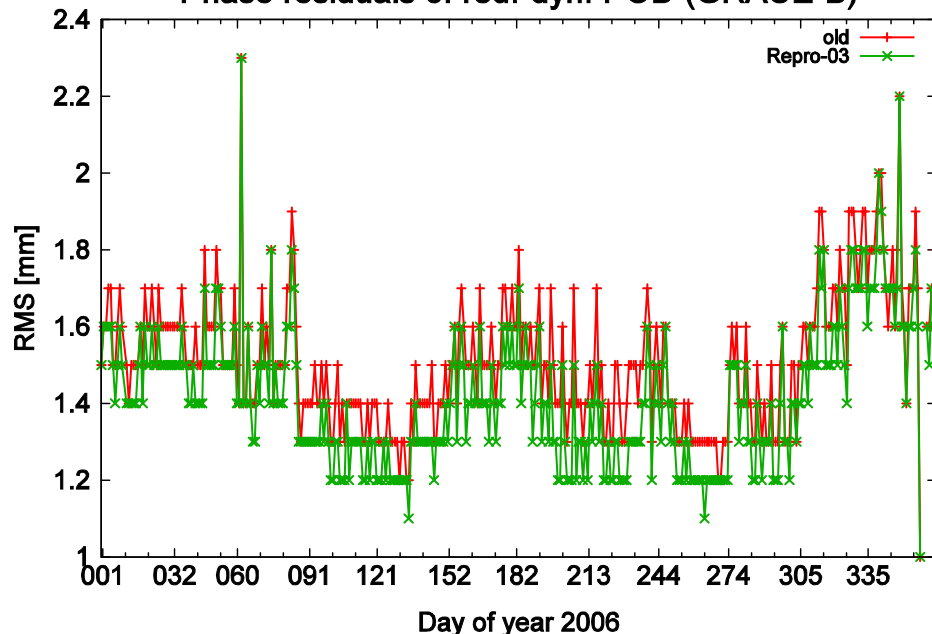


Validation by GRACE Orbit Determination

Phase residuals of red.-dyn. POD (GRACE-A)



Phase residuals of red.-dyn. POD (GRACE-B)



- Reference Frame Products:

<ftp://ftp.unibe.ch/aiub/users/susnik/.data/2006/>

<ftp://ftp.unibe.ch/aiub/users/susnik/.data/2007/>

- GRACE orbits:

- will be available by the end of the month