

EGSIEM

Title: WP 2 Gravity field analysis

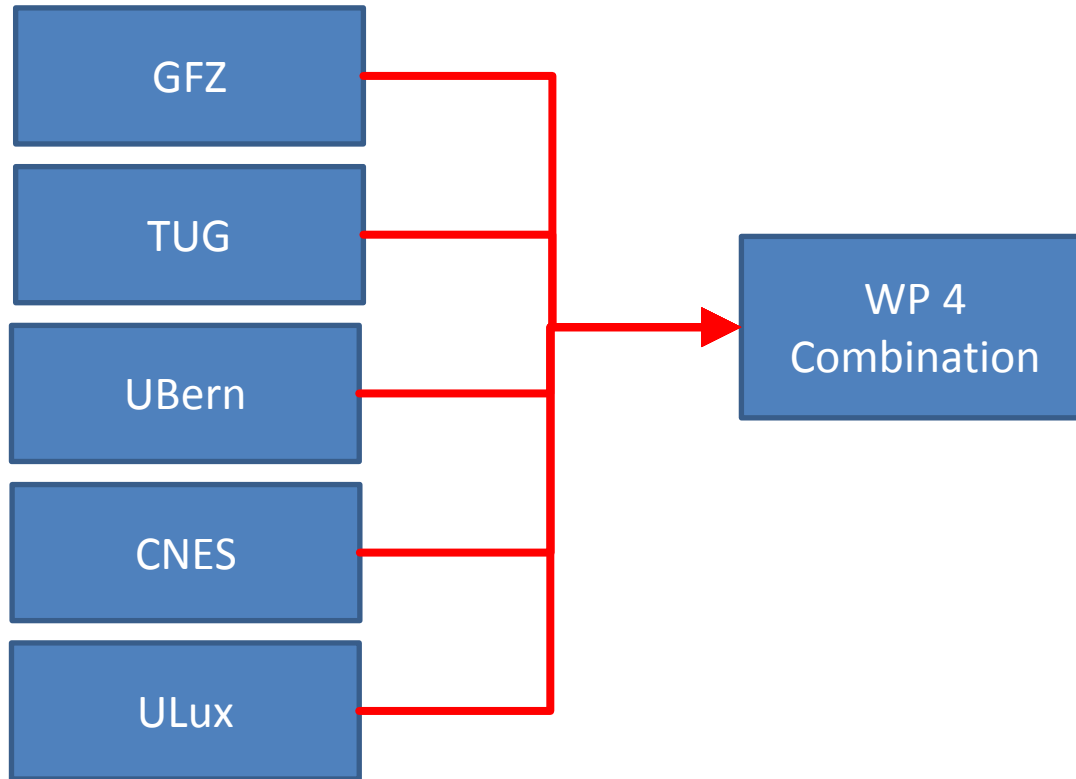
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Affiliation: TU Graz

EGSIEM Kick Off Meeting
University of Bern
January 13. – 14. 2015

WP2 Overview

Monthly GRACE solutions



WP2 Overview

- T2.1 Processing Standards and Models (UBERN, UL, GFZ, TUG, CNES) M01-M02
- T2.2 Improved processing tools (UBERN, UL, GFZ, TUG, LUH, CNES) M01-M10
- T2.3: Data analysis (UBERN, UL, GFZ, TUG, LUH, CNES) M11-M18
- T2.4: Instrumental behaviour and End-to-End Simulator (LUH, GFZ) M06-M18

T2.2 Improved processing tools
T2.3: Data analysis
(UBERN, UL, GFZ, TUG, LUH, CNES)

GRACE Level-2 Processing at GFZ

- Generation of GRACE Level-2 products based on two-step method:
 - Step 1: Calculation of precise orbits and clocks of GPS satellite constellation using GPS ground station network
 - Step 2: Adjust gravity field (and orbital/instrumental) parameters using fixed GPS reference frame from step 1
- Step 2 is based on orbit perturbation theory using the so-called “dynamic method” (also applied for CHAMP and GOCE)
- The whole processing environment required for generation of Level-2 products is consisting of the following steps:
 - Generation of precise GPS orbits & clocks
 - Preprocessing of Level-1B data (outlier detection, gap filling, reformatting etc.)
 - Definition of orbital arcs to be processed
 - Data screening and orbit determination of GRACE/GRACE-FO spacecrafts
 - Generation of normal equations
 - Solving for gravity field (and orbit/instrumental specific) parameters
 - Validation

GRACE Level-2 Processing at GFZ

- Current GFZ GRACE RL05 standards: IERS2010 conventions, IGS08 (GPS constellation), background models see table below and “GRACE Processing Standards Document for GFZ RL05 Level-2 Data”

<i>GFZ GRACE RL05 Background Models</i>	
Static Gravity Field	EIGEN-6C
Time-variable Gravity Field	Trend / Annual / Semiannual Model from EIGEN-6C till d/o 50x50
Ocean Tides	EOT11a
Atmospheric Tides S1, S2	Biancale-Bode 2003
Atmospheric and Oceanic Non-tidal Mass Variations	AOD1B RL05
Ocean Pole Tide	Desai [2002]
Solid Earth & Pole Tides	IERS2010
3 rd Body Ephemerides	JPL DE421

GRACE Level-2 Processing at GFZ

- Planned further improvements:
 - Use of latest available / best standards and background models
 - Optimized parameterization of accelerometer and K-band observations
 - Optimized screening procedure (definition of orbital arcs, elimination of observations)
 - ...
- Required auxiliary products and latencies:

• IGS raw data (<i>for GPS constellation only</i>)	1 day
• Troposphere (VMF) (<i>for GPS constellation only</i>)	1 day
• IGS final orbits (validation) (<i>for GPS constellation only</i>)	2-3 weeks
• Ionosphere (IGS daily products)	~3 weeks
• AOD1B products	~1 week
• EOP 08 C04	~6 weeks
- Required latency for L2-products (GRACE SDS requirement): 2 months

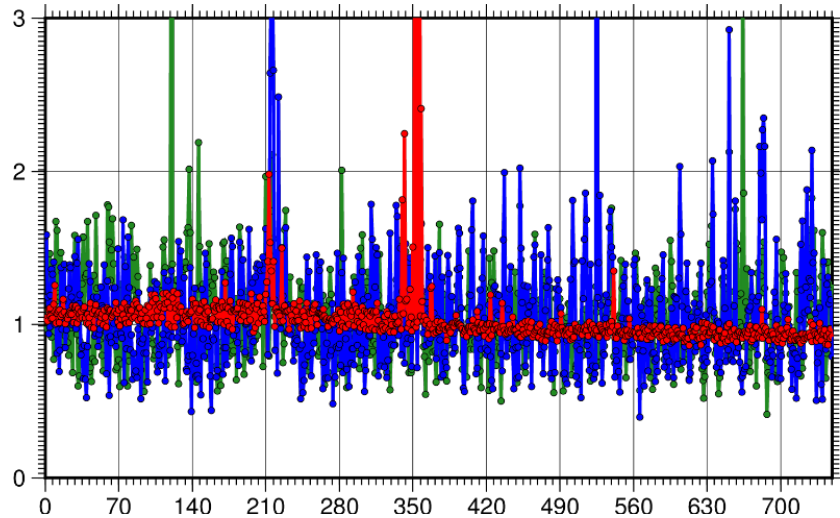
ITSG-Grace: GRACE Processing at TU Graz

- Generation of GRACE Level-2 products based on three-step method:
 - Step 1: Fixed IGS/CODE solution: GPS orbits & clocks
 - Step 2: Estimation of kinematic orbits of GRACE satellites
 - Step 3: Adjust gravity field (and orbital/instrumental) parameters

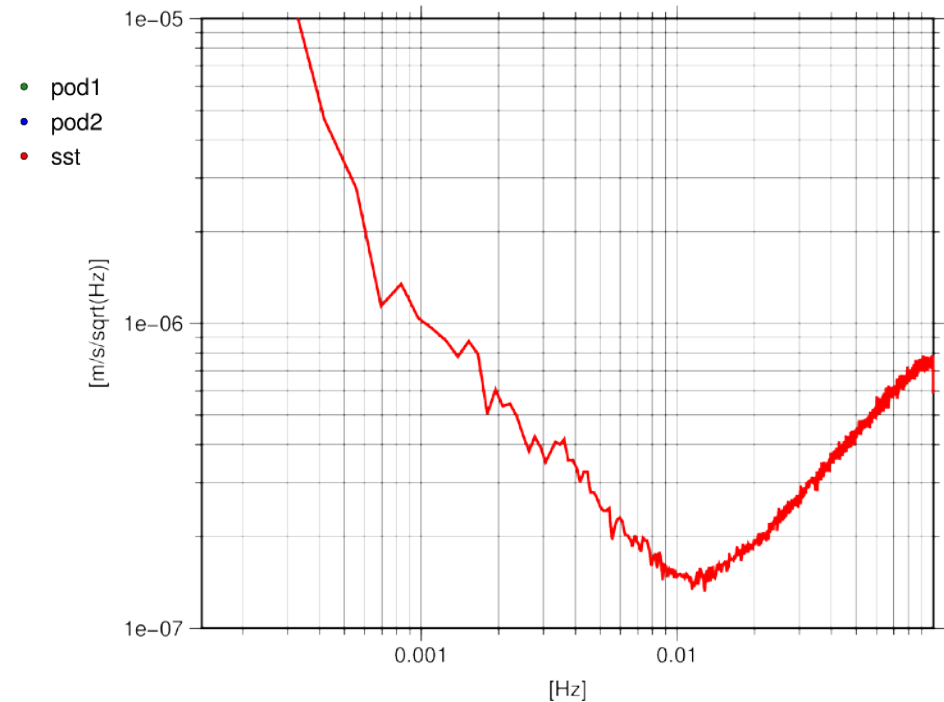
ITSG-Grace: GRACE Processing at TU Graz

- Variance component estimation for 1 hour data:
 - Accuracies/weights for each arc (KBR & orbits)
 - Empirical covariance function (via PSD)

Arc sigmas

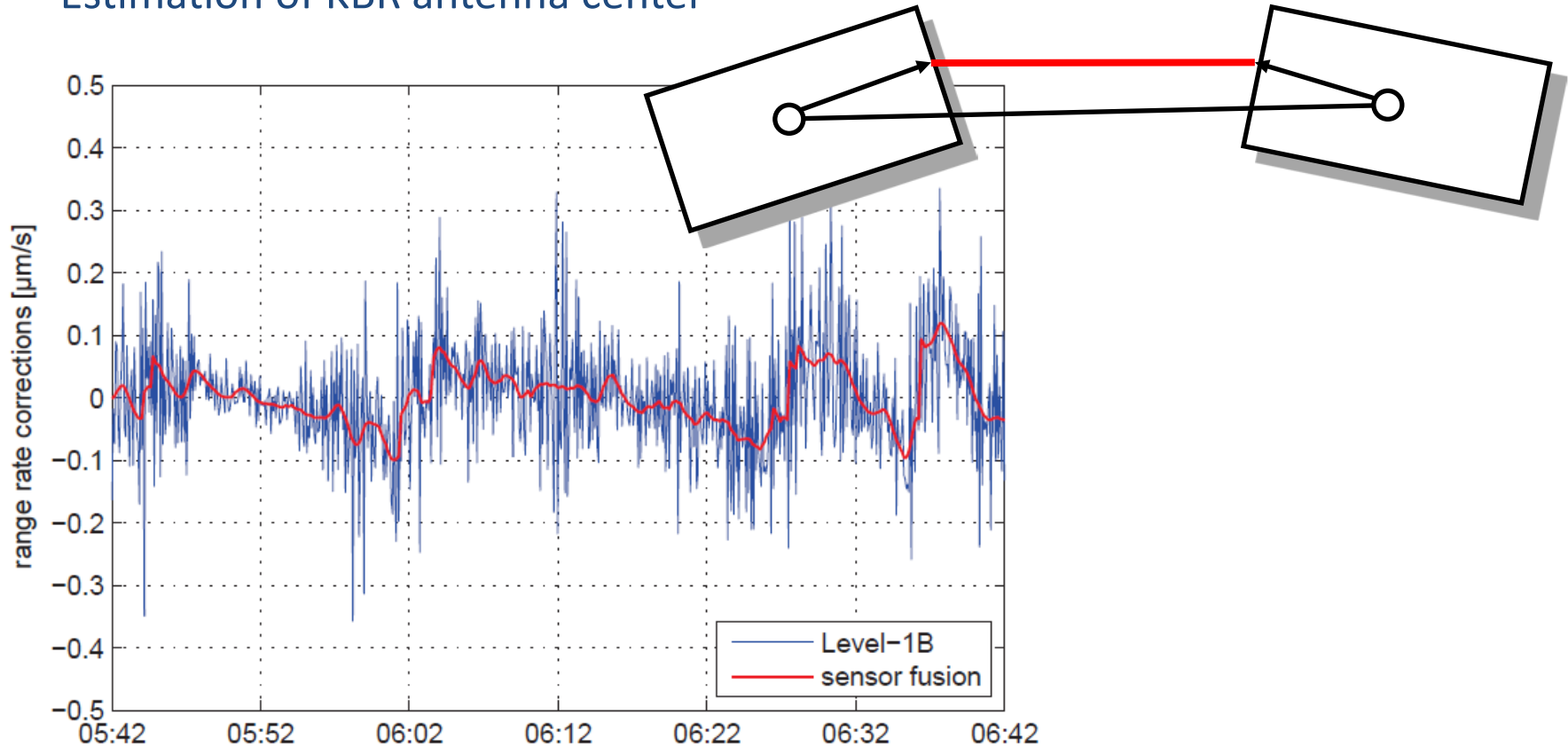


PSD of K-Band residuals



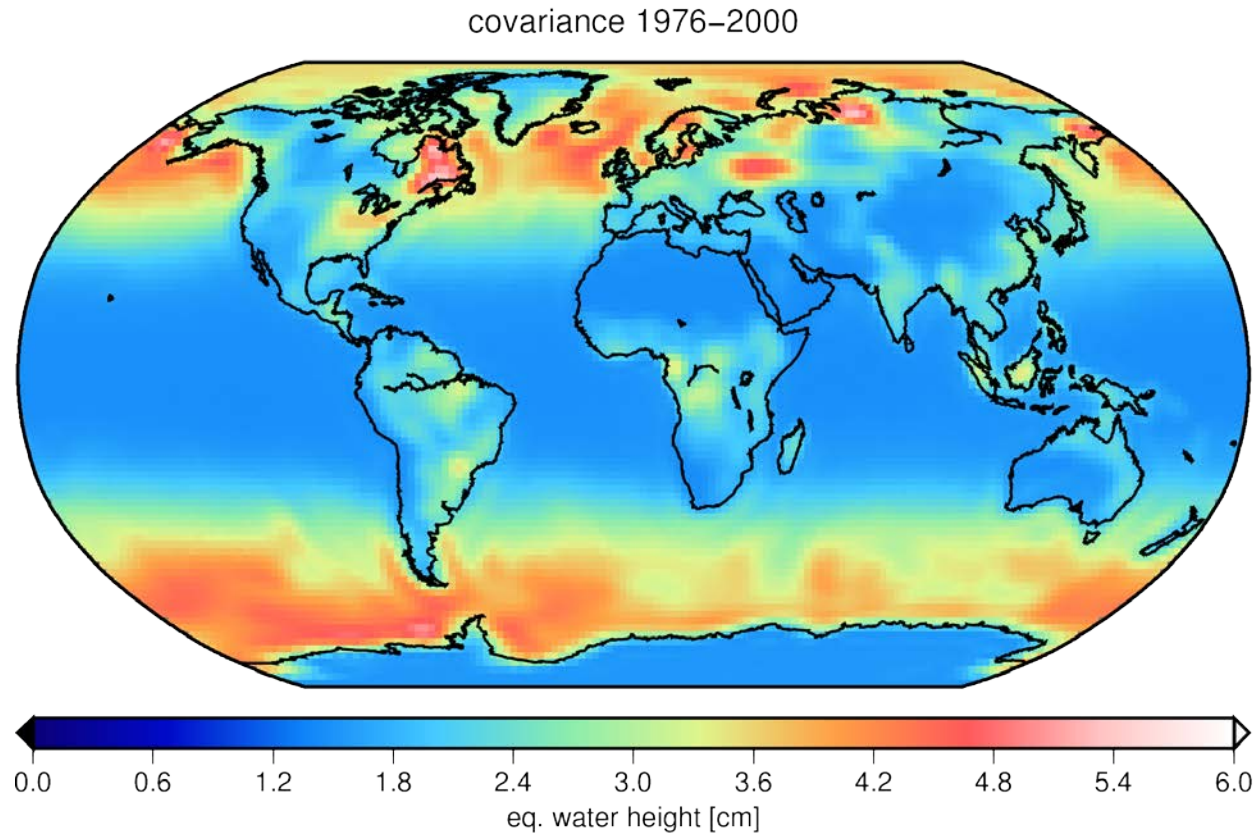
ITSG-Grace: GRACE Processing at TU Graz

- Combination of star camera data and angular variations
- Estimation of KBR antenna center



ITSG-Grace: GRACE Processing at TU Graz

- Estimation of (constrained) daily variations additional to AOD1B
- Constrained derived from geophysical models



ITSG-Grace: GRACE Processing at TU Graz

Planned further improvements

- Stochastic modelling: separation of ACC & KBR noise
- Daily variations: Use of ~3,500 daily normal to separate noise/signal via VCE
- Outlier detection

Auxiliary input:

- GNSS orbits and clock-corrections
- AOD1B products
- EOP 08 C04

CMA – Celestial Mechanics Approach (UBern)

Monthly gravity fields (90)

- | | |
|---------------|---|
| Data | <ul style="list-style-type: none">• GRACE kinematic positions• Level 1B K-band range-rates |
| Orbits | <ul style="list-style-type: none">• Initial conditions every 24h• Accelerations over 15min (constrained) |
| K-band | <ul style="list-style-type: none">• No additional parameters• Correlations not modeled |
| Accelerometer | <ul style="list-style-type: none">• Daily scale-factors in 3 directions |
| A priori | <ul style="list-style-type: none">• AIUB-GRACE03S up to degree 160• IERS 2010 solid Earth tides |
| Background | <ul style="list-style-type: none">• EOT11a ocean tides• RL05 AOD1B dealiasing |

CMA – Celestial Mechanics Approach (UBern)

- Planned improvements:

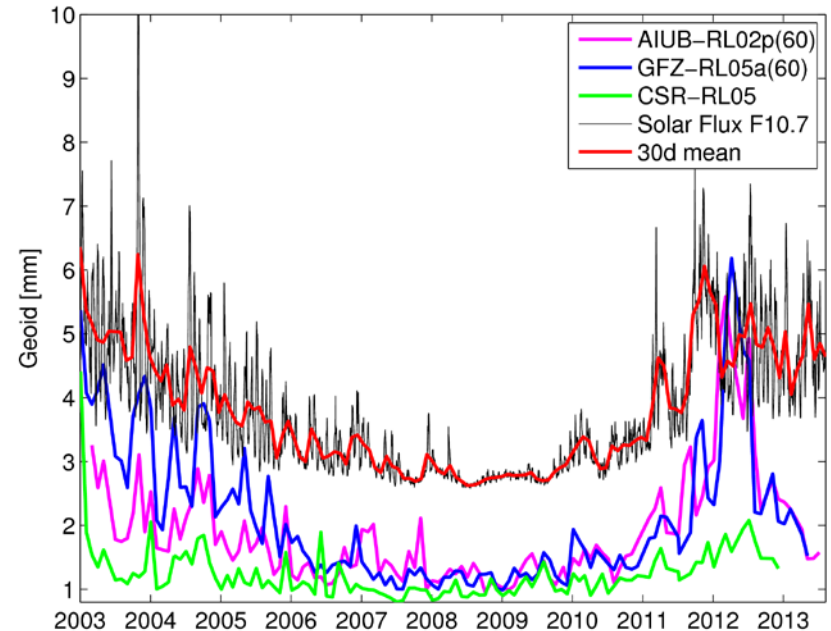
- Noise model:

- Tailored constraining of stochastic parameters
 - Stochastic covariances???
 - Down-weighting of bad observations???

- Auxiliary input:

- GNSS orbits and clock-corrections: in-house (CODE)
 - Kinematic LEO orbits: in-house
 - Atmosphere/Ocean dealiasing
 - SLR data (validation)

SHC-noise (RMS oceans)



CNES/GRGS GRACE processing: current standards of RL03

Dynamical models

Gravity	<i>EIGEN-GRGS.RL02</i> → <i>EIGEN-6S2</i>
Ocean tide	<i>FES2004 (degree 80)</i> → <i>FES2012 (Legos)</i>
Atmosphere	<i>3-D ECMWF pressure grids / 6hrs</i> → <i>ERA-interim / 3hrs</i>
Ocean mass model	<i>MOG2D (non-IB) / 6hrs</i> → <i>TUGO (Legos) / 3hrs</i>
Atmospheric tides	→ <i>Not necessary any more</i>
3 rd body	<i>Sun, Moon, 6 planets (DE405)</i>
Solid Earth tides	<i>IERS Conventions 2010</i>
Pole tides	<i>IERS Conventions 2010</i>
Non gravitational	<i>Accelerometer data (+biases and scale factors)</i>

CNES/GRGS GRACE processing: current standards of RL03

Geometrical models

SLR stations	<i>ITRF2008 coordinates</i> → <i>updated</i>
GPS	<i>IGS orbits and CODE clock</i> → <i>IGS Repro-1 orbits and clocks</i>

Other models

Hydrology	Taken into account by the a priori gravity field
Glacial Isostatic Adjustment	

Additional features

<i>Strong down-weighting and spacing of the GPS observations of GRACE</i> (wrt. RL02)
Change of inversion strategy: Choleski → <i>Truncated SVD</i>

Known problems of RL03 and planned improvements

Weakness of the sectorials

- This point was very noticeable in the unconstrained solutions that we provided to Adrian for test purposes (not usable !)
- **The cause:** a strong down-weighting of the GPS phase and range measurements on GRACE in RL03-v1.
- **Planned correction in RL03-v2:**
 - Adding Starlette and Stella to Lageos-1&2 will strengthen the very low degrees and improve some sectorials;
 - The weight of GRACE GPS phase and range data will be brought back to a reasonable value.

Known problems of RL03 and planned improvements

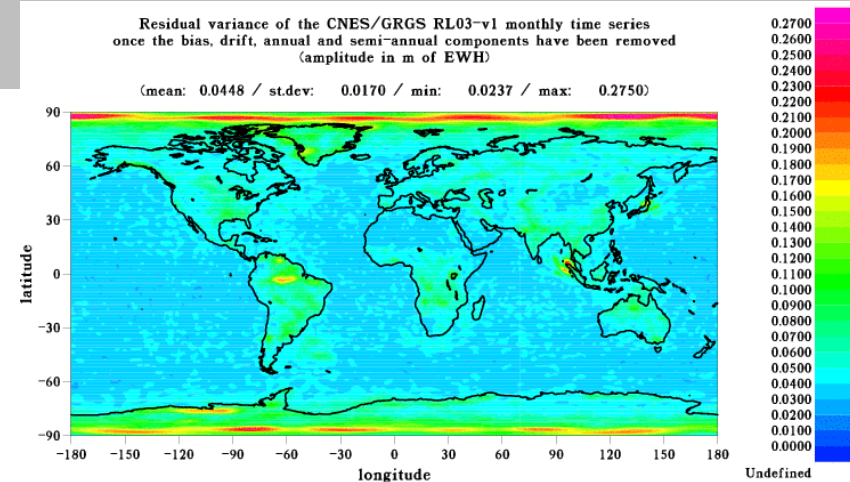
An erroneous mass signal located in two small circular rings close to the Earth's poles:

→ This has led to our recommendation **not to use** RL03-v1 above 82° latitudes North and South;

→ **The cause:** the SVD inversion strategy which does not allow for a correct solution of the low degree sectorials. These, in turn, perturb all the coefficients of a given order.

→ **Planned correction in RL03-v2:**

- Switching to a two-step approach: first a Choleski inversion of the low degrees, then injection of this solution in the normals and resolution using truncated SVD.

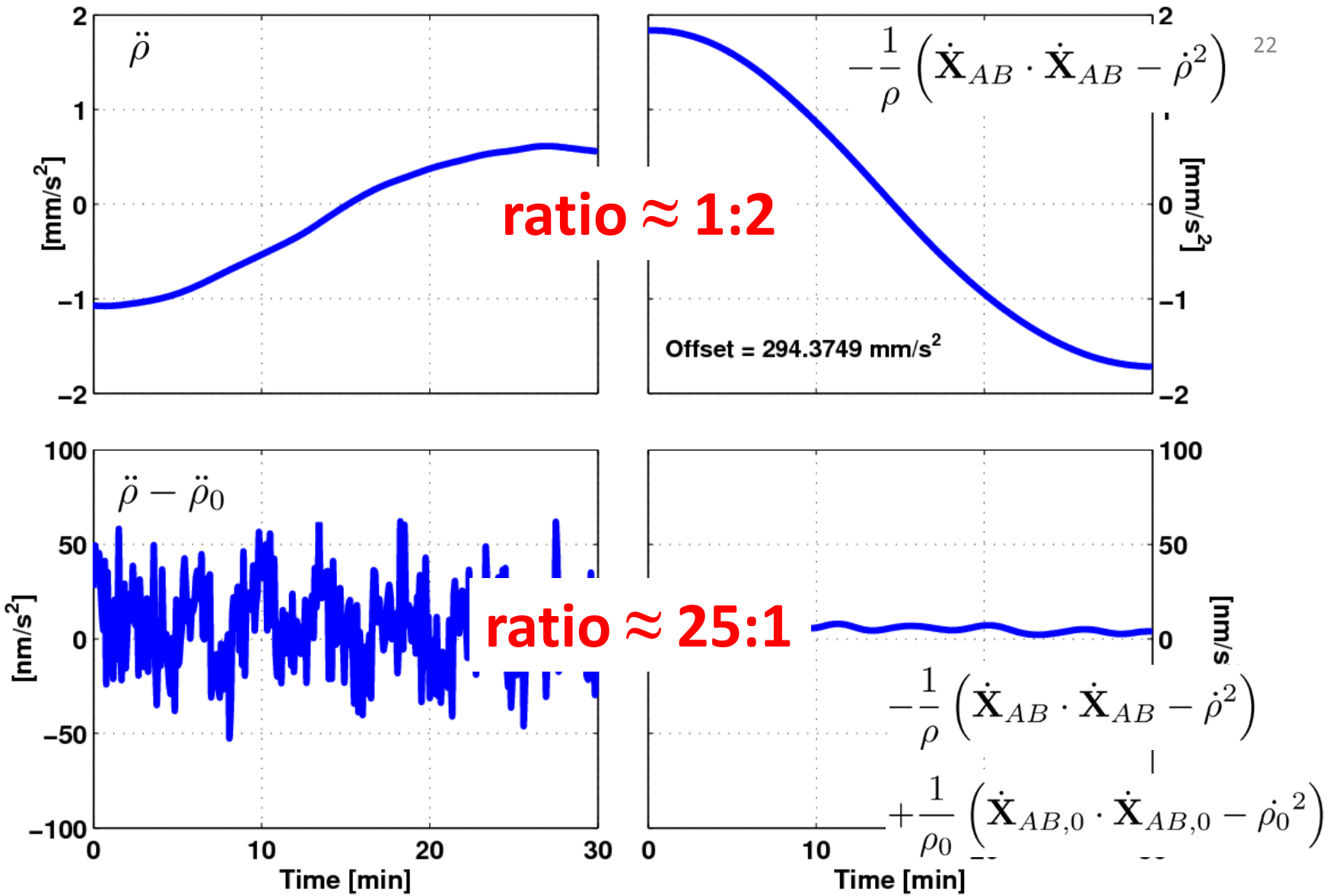


Acceleration approach (Lux)

- Many variants of the *acceleration approach* exist
- Here we refer to the acceleration approach as:

$$\nabla V \cdot \mathbf{e}_{AB} = \ddot{\rho} - \frac{1}{\rho} \left(\|\dot{\mathbf{X}}_{AB}\|^2 - \dot{\rho}^2 \right)$$

- Practical implementation:
 - Reduction to residual quantity
 - Neglecting the “GPS”-term $\nabla V \cdot \mathbf{e}_{AB} - \nabla V_0 \cdot \mathbf{e}_{AB,0} \approx \ddot{\rho} - \ddot{\rho}_0$
 - Iteration (high numerical effort)
 - currently solutions are not competitive yet (primarily limited by the implementation of the orbit integration)



Future improvements

- Implementation:

- fix problems with orbit integration
- add stochastic pulses/constant accelerations
- fix parameterization
- migration to cluster

- Methodological:

considering the “GPS” term via variational equation

$$\ddot{\rho} - \ddot{\rho}_0 \approx \nabla V \cdot \mathbf{e}_{AB} - \nabla V_0 \cdot \mathbf{e}_{AB,0} + \frac{\partial}{\partial \bar{K}_{lm}} \left[\frac{1}{\rho} \left(\|\dot{\mathbf{X}}_{AB}\|^2 - \dot{\rho}^2 \right) \right] \Delta \bar{K}_{lm}$$

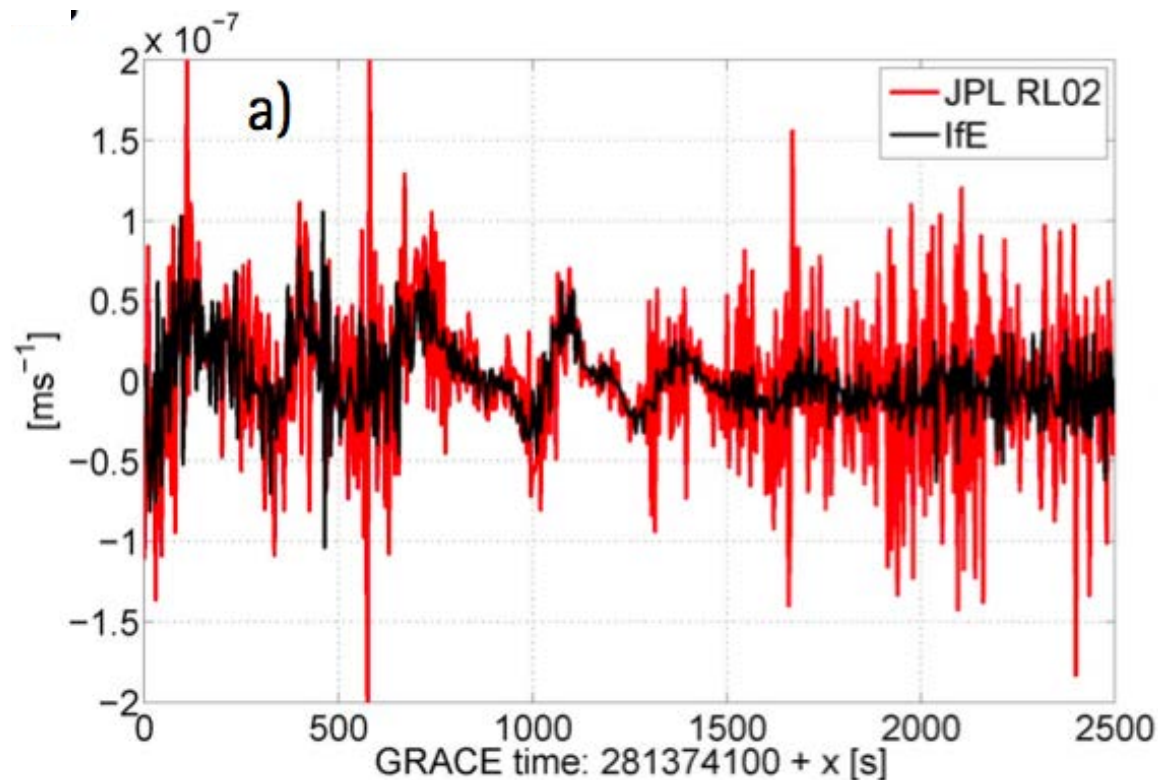
T2.4: Instrumental behavior and End-to-End Simulator

L1B Sensor data analysis (LUH)

- for reprocessing (T2.3)
- for noise time series for E2E simulation (T2.4)
- main topics:
 - accelerometer data
 - attitude data
 - LRI data

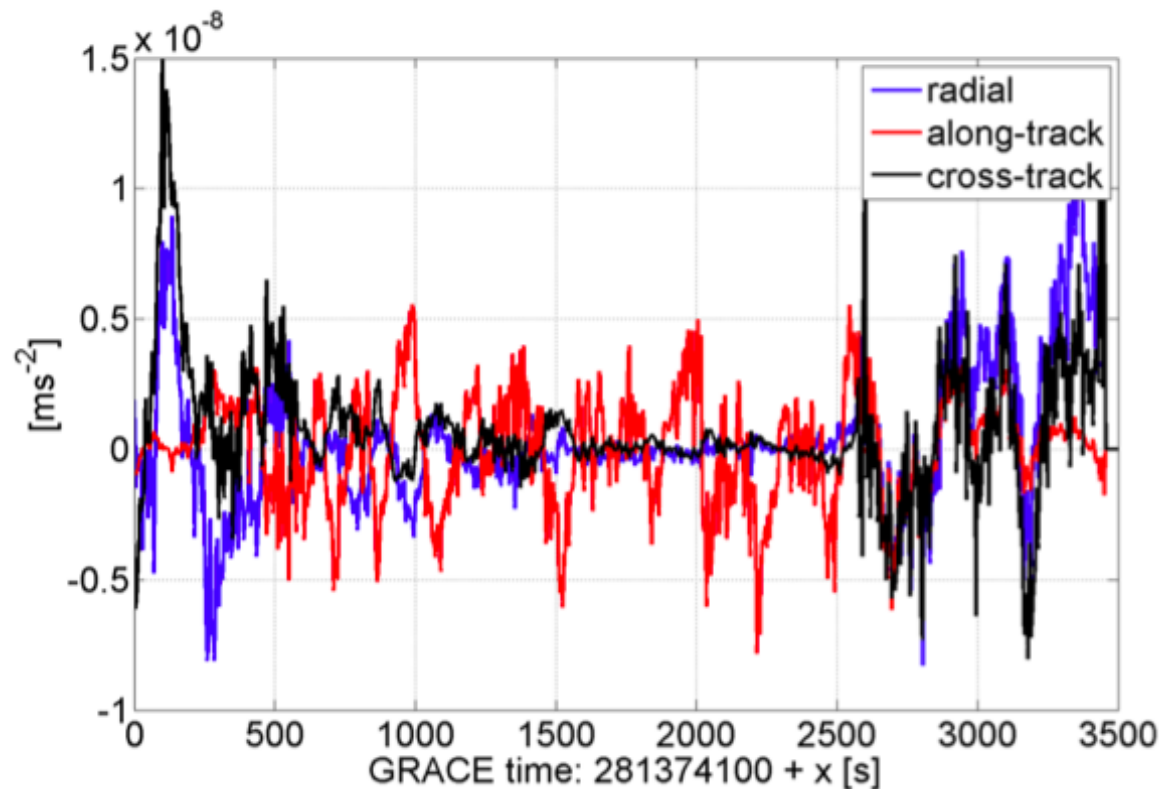
Sensor data analysis: Attitude

- Star camera common frame reprocessing
– effect on inter-satellite range-rate



Sensor data analysis: Attitude

- Star camera common frame reprocessing
– effect on non-gravitational accelerations

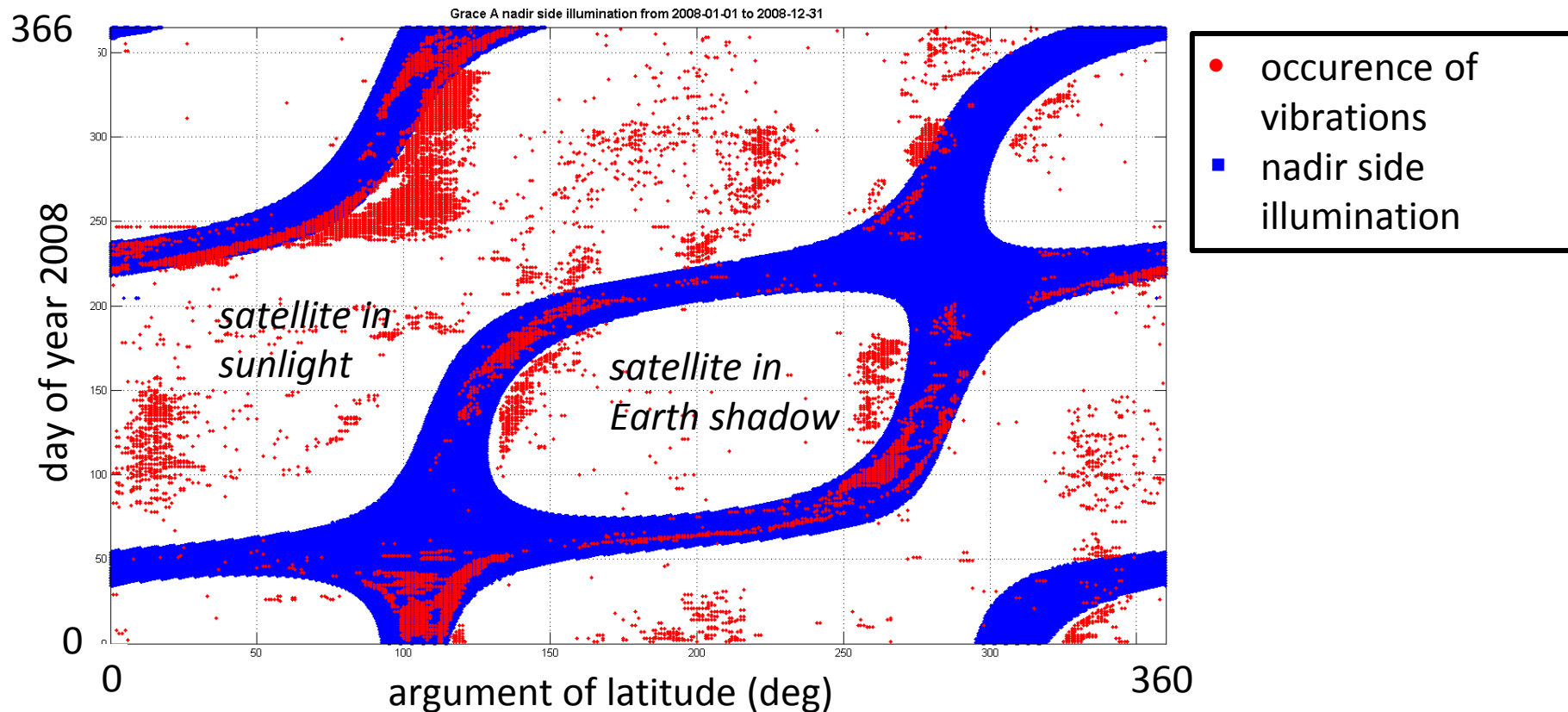


Sensor data analysis: Attitude

- attitude related errors
 - measurement errors
 - alignment errors
 - KBR phase center errors
- research on separation ongoing
- coupled with stochastic modeling
- options for sensor fusion

Sensor data analysis: Accelerometer

- environmental disturbances: twangs, vibrations, heater switching spikes, MTQ spikes



Sensor data analysis: Accelerometer

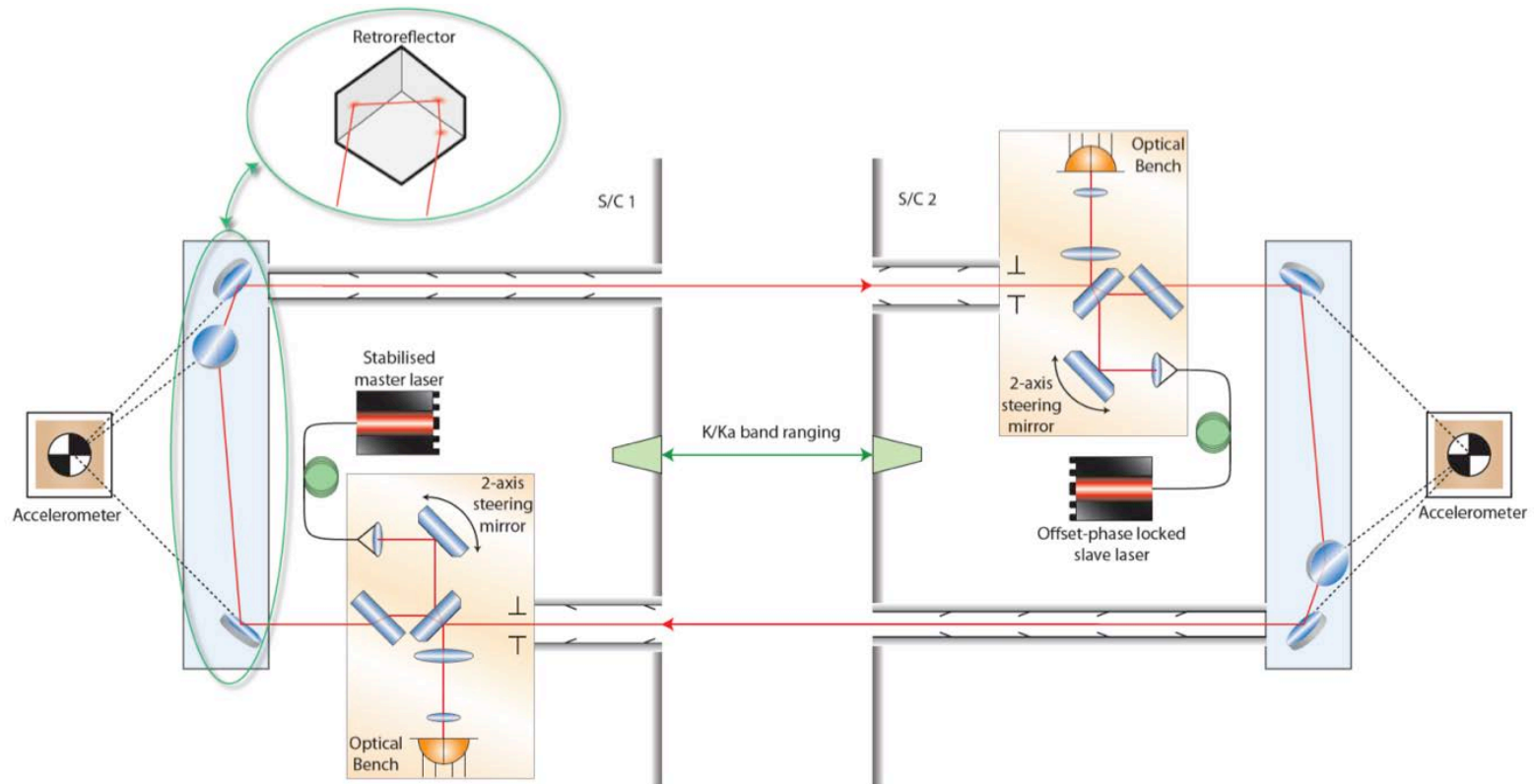
- environmental disturbances
 - re-processing with models for heater switching spikes, MTQ spikes, (twangs tbc), topic of SFB1128 geo-Q
 - noise time series

T2.3 Re-processing of L1B data (M11-M18)

- star camera reprocessing
 - common frame
 - ...
- accelerometer
 - reductions for heater switching, MTQ spikes
 - twangs tbc

Sensor data analysis: LRI

- noise modeling



T2.4 Noise time series for E2E simulation (M06 – M18)

- harmonize input with planned E2E rounds and scenarios
- attitude noise time series
 - propagated to range-rate (line of sight)
 - propagated to non-gravitational accelerations (3 axes)
- environmental accelerometer noise time series (3 axes)
- LRI noise time series
- investigate interaction with other effects
 - background model errors
 - orbit parameterization
 - ...

GFZ E2E Simulator: General

- Based on GFZ EPOS (Earth Parameter and Orbit System) S/W used for real data analysis e.g. of gravity (CHAMP, GRACE, GOCE), SLR or altimetry missions
- Has been used for various simulation studies of NGGMs, e.g.
 - ESA study NG2 (2011) „Assessment of a Next Generation Gravity Mission to Monitor the Variations of Earth’s Gravity Field“ by Airbus, GFZ and University of Bonn
 - BMBF study (2013) “Future Gravity Field Satellite Missions” within Geotechnologien Program “Observation of the System Earth from Space - CHAMP, GRACE, GOCE and future missions” (GFZ, industry and various D universities)
 - DLR study (2014) „Next Generation Gravity Mission – Deutschland“ (NGGM-D) (TU Munich, GFZ, industry, various D universities)

GFZ E2E Simulator: 5-Step Approach

- Step 1 (Constellation)
 - Define number of pairs, orbital parameters, mission duration etc.
- Step 2 (Simulation)
 - Define a-priori („true world“) background models (static gravity field, ocean tides, non-tidal mass variations (AOHIS))
 - Simulate daily white noise HL-SST (GPS code and phase)
 - Simulate daily geometric LL-SST (MWI or LRI range rate)
 - Simulate daily non-gravitational forces (from air drag, solar radiation and albedo models) and transform to accelerometer data (ACC)
- Step 3 (Noise)
 - Add realistic (colored) noise to observations (LL-SST, ACC) provided by University Hannover

GFZ E2E Simulator: 5-Step Approach

- Step 4 (Recovery)
 - Exchange „true world“ background models from step 1 by models which describe „best knowledge“ uncertainties and perform monthly gravity field determination using noisy observations from step 3
 - During recovery step similar parameters are adjusted as in real case scenarios (initial state vector, 3h ACC bias and scales, MWI/LRI parameters etc.)
- Step 5 (Analysis)
 - Compare for all months of missions duration „recovered“ and „true world“ monthly models in the spectral and spatial domain e.g. for independently calculated LRI or MWI cases

GFZ E2E Simulator: Example Background Models

- Step 2 (Simulation)
 - static gravity model (EIGEN-GL04C (Förste et al. 2008) up to degree and order 100
 - Sun and Moon ephemerides (DE405, Standish 1998),
 - ocean tides (8 main constituents Q1, O1, P1, K1, N2, M2, S2 and K2 of EOT08a, Savcenko et al. 2008)
 - non-tidal short-term mass variations for the atmosphere, oceans, hydrology, ice and solid Earth (AOHIS, Dobslaw et al. (2014))
- Step 4 (Recovery)
 - static gravity model EGM96 (Lemoine et al. 1998) up to d/o 100
 - Ocean tides (8 main constituents substituted by GOT4.7 (Ray 2008)
 - errors to the AOHIS model as described in Dobslaw et al. (2014)

GFZ E2E Simulator: Example NGGM-D

Initial baseline scenario

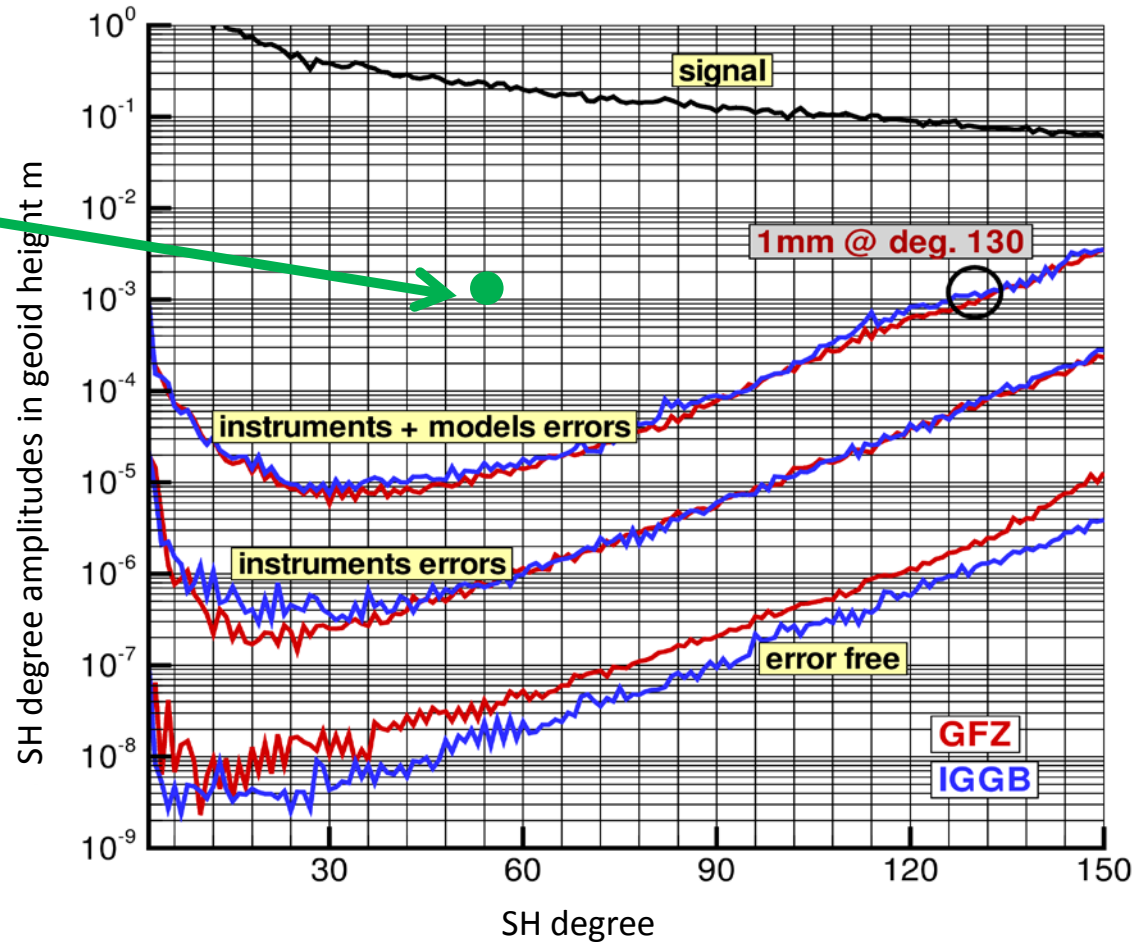
- Double pair constellation (“Bender”)
- Near polar orbit & inclined orbit @ $h=420\text{km}$ (2nd pair @ 70°)
- Satellite distance: 100 km
- Maximum spherical harmonic degree: 150 (133 km)
- Monthly period
- SST noise: 10^{-8} m (MBW) (25 nm vs 2000 nm (GRACE))
- ACC noise: 10^{-11} m/s² (MBW) (10 times GRACE)

Requirements for monthly geoid error (w/o postprocessing)

- 1mm geoid (1 cm EWH) @ 200km (minimum, 10 times better than GRACE)
- 1mm geoid (1 cm EWH) @ 150km (optimum)

GFZ E2E Simulator: Example NGGM-D

GFZ GRACE RL05a:
1mm @ deg 50



- Consistent independent GFZ and IGG Bonn solutions
- Requirements fulfilled!

T2.1 Processing Standards and Models

T2.1 Processing Standards and Models

Goal: Combined gravity field solution based on normal equation level (WP4)

We should define only minimum requirements

Allow much freedom in processing as possible

Normal equations:

- Unconstrained
- Satellite parameters pre-eliminated
- Background models restored (static field, ...)
- GRACE-only solutions (Combination of SLR on normal equation level)
- Format: SINEX?

T2.1.2 Reference system

- Center of mass (CM) :
Degree 1 coefficients of all background must be zero
- Tidal system:
Zero tide (as recommended by IERS2010)
- Reference system
ITRF, no further recommendation needed?
(GRACE horizontal resolution $\gg 10$ km)

T2.1.1 Background models

- Static field, trend, annual, semiannual: Different fields
- Third body forces (sun, moon, planets): JPL DE421
- Solid earth tides: IERS 2003/2010
- Ocean tides: EOT11a / FES2012 / ...
(incl. equilibrium tides: Om1, Om2, Sa, Ssa, ...)
- Pole tides: IERS 2010 (w/o equation 6.4)
- Ocean pole tides: IERS 2010
- Atmospheric tides (S1, S2) Bode-Biancale 2003 / ...
- Dealiasing: AOD1B RL05 / MOG2D / ...
(Atmospheric S1/S2 removed)

T2.1.1 Background models

Different purposes

- Taylor point of Linearization
 - E.g. Static field, trend, annual, semiannual
 - Should be added back to normal equations
- Remove high frequent signals (< 1 month) for dealiasing
 - True signal is unknown: e.g. AOD1B vs. MOG2D
 - Consistency not possible and not meaningful
- Signal separation (≥ 1 month)
 - Depends on the application
 - GRACE observes the full signal, but at the moment:
 - Tidal effects removed
 - Atmospheric and ocean signals (some parts) are removed
 - Hydrology and Ice included

Tides: Standards required for signals > 1 month)

Add back monthly mean of dealiasing products before combination

Problem: Some of the background models are use for multiple purposes

T2.1.3 Constraints

Constraints / filtering / regularization / a priori information:

- Helps to improve the solution, BUT:
- Damps always the noise and the signal
- Smooth transition between too much and too less regularization
- Depends on the signal to noise ratio

=> Regularization not allowed before combination

T2.1.3 Constraints / Empirical parameters

Constraints:

Only constraints for pre-eliminated parameters are allowed

Possible empirical parametrization (pre-eliminated):

- Accelerometer (Mandatory: scale and bias estimation)
- KBR parametrization (bias, trend ...)
- KBR antenna center variations
- Empirical orbits parameters, stochastic pulses, ...
- Full noise variance-covariance matrix

T2.1.4 GPS observations / kinematic orbits

- Reduced-dynamic orbits:
 - Not allowed in observation vector
 - > possible bias towards reference gravity field
- Further standards needed here?
(IGS standards?)

T2.1 Processing Standards and Models

Goal: Combined gravity field solution based on normal equation level (WP4)

We should define only minimum requirements

Allow much freedom in processing as possible

Normal equations:

- Unconstrained
- Satellite parameters pre-eliminated
- Monthly mean of background models restored
 except tidal effects: Earth tides, long periodic ocean tides, pole tides
- GRACE-only solutions
- Format: SINEX?

Standards: overview

Approaches may differ by:

- Philosophy
- Input data
- Orbit dynamics
 - conventions
 - background models
- Earth kinematics (orientation, reference frame)
- Satellite kinematics (orientation, geom. biases)

Standards: methods, data, solutions

GFZ-RL05a (GFZ L2 Proc. Std. Doc.)	CSR-RL05 (UTCSR L2 Proc. Std. Doc.)	AIUB-RL02
Method: dynamic approach 1h ACC biases no emp. ACC fixed relative weighting	Method: dynamic approach ACC biases and scales no emp. ACC optimal KRR-weighting	Method: dynamic approach 15 min. emp. ACC, constr. Const. and 1/rev. emp. ACC fixed relative weighting
Data: ACC 5s, ATT, range-rates 5s, no corr.	Data: ACC 1s, ATT, range-rates 5s, no corr.	Data: ACC 1s, ATT, range-rates 5s, no corr.
GPS code/phase: cutoff 10° undifferenced Offset: 0/0/-444mm, PCVs	GPS code/phase: double diff., 2 minutes Offset: 0/0/-490mm, PCVs	Kin. orbits.: GPS undifferenced, 30 s Offset: 0/0/-452, PCVs
Solutions: tide-free monthly (90), weekly (30) mostly unconstrained	Solutions: zero-tide monthly (96) unconstrained	Solutions: tide-free monthly (90) unconstrained

Standards: background models

GFZ-RL05a	CSR-RL05	AIUB-RL02
<p>A priori: EIGEN-6C (200) combined trends + per. var. (50) $a_E=6378136.46\text{m}$</p>	<p>A priori: GIF48 (360) combined trends not applied $a_E=6378136.3\text{m}$</p>	<p>A priori: AIUB-GRACE03S (160), satellite only tr.+ per. var. (30) not appl. $a_E=6378137.0\text{m}$</p>
<p>Ocean tides: EOT11A (80) M_{tm}, M_{sqm}: FES2004 $\Omega_{1,2}, S_{a,sa}$: HW95 admittances interpol.</p>	<p>Ocean tides: GOT4.8 (180) M_{tm}, M_{sm}: FES2004 M_m, M_f: Egbert & Ray(2003) admittances interpol.</p>	<p>Ocean tides: EOT11a(100) M_{tm}, M_{sqm}: FES2004 $\Omega_{1,2}, S_{a,sa}$: HW95 admittances interpol.</p>
<p>Non-tidal variations: AOD1B-RL05 (100) S2 removed</p>	<p>Non-tidal variations: AOD1B-RL05 (100) S2 removed</p>	<p>Non-tidal variations: AOD1B-RL05 (100)</p>
<p>Atmosphere tides: Bode & Biancale (2006) S1 and S2 (8/5)</p>	<p>Atmosphere tides: Ray & Ponte (2003)</p>	<p>Atmosphere tides: none</p>

Standards: orbit dynamics

GFZ-RL05a	CSR-RL05	AIUB-RL02
Solid Earth tides: IERS2010 Planetary eph.: DE 421	Solid Earth tides: IERS2010 Planetary eph.: DE 405	Solid Earth tides: IERS2010 Planetary eph.: DE 405
Pole tide: IERS2010 IERS EOP 08 C04	Pole tide: IERS2003 IERS C04	Pole tide: IERS2010
Ocean pole tide: Desai (30)	Ocean pole tide: Desai (100)	Ocean pole tide: Desai (100)
3rd bodies: point masses Sun, moon, 5 planets	3rd bodies: point masses Sun, moon, planets	3rd bodies: point masses Sun, moon, planets
Relativistic effects: IERS2010, incl. Lense- Thirring and de Sitter	Relativistic effects: IERS2010	Relativistic effects: IERS2010

Standards: Earth kinematics

GFZ-RL05a	CSR-RL05	AIUB-RL02
Reference frame: J2000.0, IGS08	Reference frame: J2000.0, IGS08	Reference frame: J2000.0, IGS08
Precession and nutation: IERS2010 (IAU2006/2000)	Precession and nutation: IERS2003 (IAU2000A)	Precession and nutation: IERS2010 (IAU2006/2000)
Sidereal rotation: ERA: IERS2010	Sidereal rotation: GMST: IERS2003 diurnal tide corr.: IERS1996	Sidereal rotation: IERS2010
Polar motion: IERS2010 including s' (TEO)	Polar motion: IERS2003 ocean tidal var.: IERS1996	Polar motion: IERS2010

Standards within the IGS

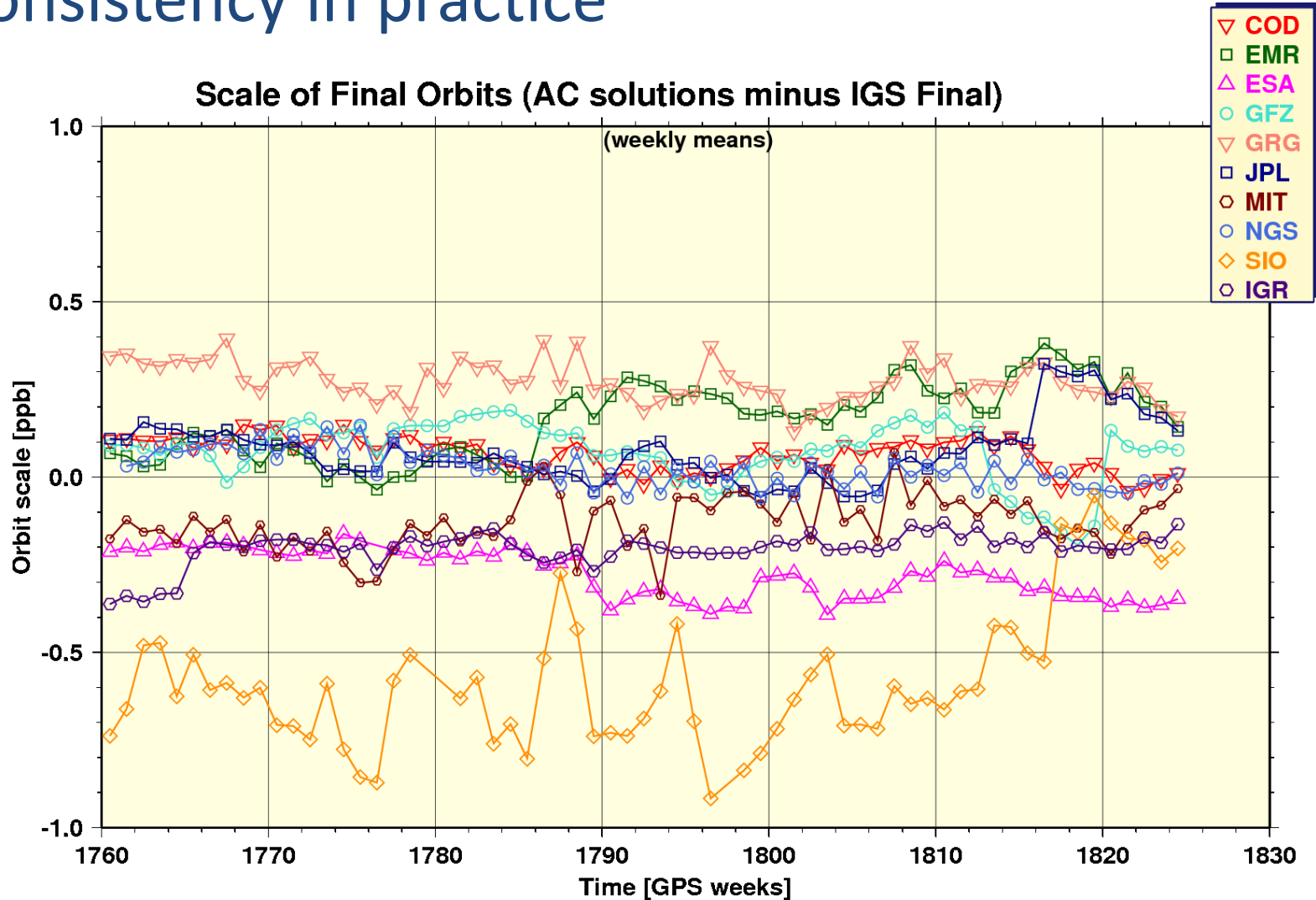
Standards within the IGS

The IGS analysis centers are asked to provide solutions

- following the **latest IERS convention** and
- referring to the **most recent ITRF**

to obtain a **consistent set of solutions** allowing a reasonable combination.

Consistency in practice



Important is to know, what people are doing:

- Exchange of most important information:

	A	B	C	D	E	F	G	H
1			CODE (COF)	EMR	ESA	GFZ	GRG	JPL
51								
52	<i>Tropospheric Delay</i>	a priori met source & mapping coeffs?	VMF1 mapping fnc coefficients interpolated from 6-hr grids	VMF1 mapping fnc coefficients interpolated from grids	GPT for P, T, RH	GPT2 for P, T, RH & VMF1 mapping fnc coefficients (wk 1758)	GPT2 for P, T, RH & VMF1 mapping fnc coefficients	GPT2 for P, T, RH & VMF1 mapping fnc coefficients
53		a priori zenith delay	ECMWF-based ZDD	VMF1 grid files for ZDD & ZWD	ZDD & ZWD computed with Saastamoinen (1972) model	ZDD & ZWD computed with Saastamoinen (1972) model	ZDD & ZWD computed with Saastamoinen (1972) model	GPT2 for P, T, RH & VMF1 mapping fnc coefficients
54		Mapping of a priori zenith delay to line-of-sight	slant dry + wet delays mapped with VMF1_HT	slant dry + wet delays mapped with GMF	slant dry + wet delays mapped with GMF	slant dry + wet delays mapped with VMF1_HT model using GPT2 mapping fnc coeffs	slant dry + wet delays mapped with GMF	slant dry + wet delays mapped with GMF
55		Mapping function used for ZD adjustment?	VMF1_HT wet mapping fnc (VMF1W); ZWD estimated in 2-hr piece-wise linear steps for each station; loosely	GMF wet mapping fnc; ZWD estimated at each epoch as random walk with process noise = 3.0	GMF wet mapping fnc; ZWD estimated in 1-hr steps for each station	VMF1_HT wet mapping fnc (VMF1W); ZWD estimated in 1-hr steps for each station	VMF1 wet mapping fnc; ZWD estimated in 2-hr steps for each station	GPT2 wet mapping fnc; ZWD estimated at each epoch as random walk with process noise = 3.0

Important is to know, what people are doing:

- Differences are indicated and discussed:

	A	B	C	D	E	F	G	H
1			CODE (COF)	EMR	ESA	GFZ	GRG	JPL
58								
59	<i>Ionospheric Delay</i>	1st-order effect	eliminated by L1 & L2 linear combination	eliminated by L1 & L2 linear combination	eliminated by L1 & L2 linear combination	eliminated by L1 & L2 linear combination	eliminated by L1 & L2 linear combination	eliminate L2 linear c
60		2nd-order	IERS 2010 & IGRF11	IERS 2010 & IGRF11	none	IERS 2010 & IGRF11 (began 1813) (TEC from CODE-iono grids)	IERS2010 & IGRF2011 (TEC from IGS/IGR-iono grids)	Applied Fernandes before 19 for 1999
61		3rd-order	IERS 2010 & IGRF11	none	none	none	none	nc
62								
63		Adjustment method	weighted least-squares	square-root information filter (SRIF)	Bayesian weighted least squares	least-squares adjustment according to Ge et al. (2006)	weighted least-squares	stochastic filter/sr implementation square

Usual practice in the IGS:

- The coordinator keeps track on the changes at all analysis centers.
 - Significant differences are discussed between the analysis centers at regular meetings.
 - If a model change is an improvement all other groups become aware of this and follow the development (or adapt it).
- Keyword: «Friendly competition»

Achievements in the IGS

A banner for the IGS Workshop 2014. The background is light blue with a colorful, abstract graphic on the left side consisting of overlapping, jagged shapes in shades of purple, pink, orange, and yellow. In the center, there is a white logo consisting of six squares and two dots. Below the logo, the text 'IGS' is written in a large, white, sans-serif font. Underneath 'IGS', the following text is displayed in a smaller, white, sans-serif font: 'Workshop 2014', 'June 23-27', and 'Pasadena, California, USA'. At the bottom of the banner, the text 'Celebrating 20 Years of Service' is written in a large, white, sans-serif font, followed by '1994 ☆ 2014' in a smaller font.

IGS

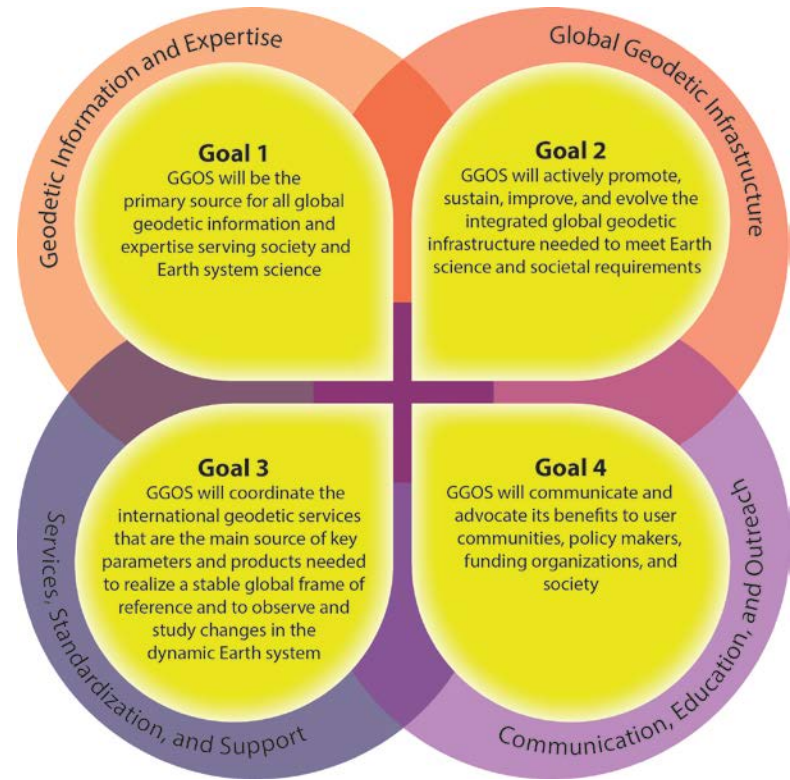
Workshop 2014
June 23-27
Pasadena, California, USA

Celebrating 20 Years of Service
1994 ☆ 2014

TUM-IAPG Expertise relevant to EGSiEM

Products & Standards

- Together with TUM-DGFI, IAPG is operating the GGOS Bureau for Products and Standards (GGOS-BPS).
- GGOS is an initiative of IAG with 4 major goals.



Products & Standards

- GGOS-BPS shall act as „**contact & coordinating point**“ regarding homogenization of standards and IAG/GGOS products.
- IAG Services shall keep their full responsibility and visibility.

Objective 3-1
Standardization

Goal 3
GGOS will coordinate the international geodetic services that are the main source of key parameters and products needed to realize a stable global frame of reference and to observe and study changes in the dynamic Earth system

Objective 3-2
Coordination and Development of IAG Services

Products & Standards

GGOS-BPS Tasks:

- Ensure consistent standards across all IAG components. „Inventory of Standards and Conventions“ (Issue 1 under review).
- Identify user needs and requirements for products.
- Initiate steps and procedures for the development of new and integrated products.
- GGOS-BPS includes GGOS-WG „Contributions to Earth SystemModelling“.

Proposal for Standards

Proposal

- Action Item (End of January):
 - GFZ, UBERN, (CSR): Review the comparison table
 - TUG, CNES, UL: Fill out the comparison table
 - UBERN: Summarize KO outcome of Standards harmonization
 - UBERN, UL, GFZ, TUG, CNES: Provide input to other processing details (parametrization, ...) identified at KO
- Action Item (Mid of February):
 - UBERN: Create the EGSiem Standards document (inspired by the GRACE Standards from GFZ/CSR) and point out commonalities/differences between individual EGSiem ACs
 - TELECON UBERN, GFZ, TUG, CNES, UL
- Action Item (End of February):
 - UBERN, UL, GFZ, TUG, CNES: Review and Finalization of EGSiem Standards Document